

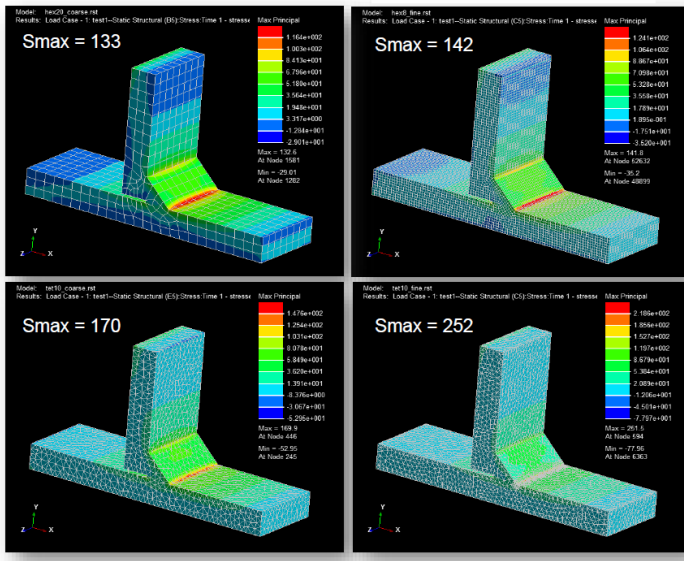


Prencia Technology Days

Virtual Seminar Series – Tuesday 2, 9, 16 November

Defining 'Stress'

"Consistent inaccuracy beats inconsistent accuracy!"



Introducing Fatigue Testing and Fatigue Simulation of Structural Joints

Presented by Dr. Andrew Halfpenny
Director of Technology
Prencia (HBK)

Tuesday, November 2, 2021

Introducing Fatigue Testing and Fatigue Simulation of Structural Joints

Abstract

A wide range of joints are used to assemble structures including welds, rivets, bolts, adhesives, between similar and dis-similar materials - metals, polymers and composites. These joints distribute load into or through the structure, and when these loads vary during operational use they introduce fatigue cycling at and close to the joint. The local geometry detail at the joint acts as a stress raiser reducing the fatigue strength at the joint. Any evaluation of the durability of the structure must therefore place a high priority on a fatigue assessment of the joints.

This presentation introduces fatigue of joints from their fatigue testing and supporting finite element analyses, through to their application in fatigue simulation analysis. Examples include welded joints (thin, thick, spot, seam, etc), brazed joints, bolted joints, adhesive joints, and self-piercing-rivet joints.

About the presenter, Dr. Andrew Halfpenny, Chief Technologist, Prenscia (HBK)

Dr. Halfpenny has a PhD in Mechanical Engineering from University College London (UCL) and a Masters' in Civil and Structural Engineering. With over 25 years of experience in structural dynamics, vibration, fatigue and fracture, he has introduced many new technologies to the industry including: FE-based vibration fatigue analysis, crack growth simulation and accelerated vibration testing. He holds a European patent for the 'Damage monitoring tag' and developed the new vibration standard used for qualifying UK military helicopters.

He has worked in consultancy with customers across the UK, Europe, Americas and the Far East, and has written publications on Fatigue, Digital Signal Processing and Structural Health Monitoring. He sits on the NAFEMS committee for Dynamic Testing and is a guest lecturer on structural dynamics with The University of Sheffield.

Introducing: Fatigue Testing and Simulation of Structural Joints

Dr Andrew Halfpenny

Director of Technology – nCode Products



Agenda

1. Joint Classification

- Discrete (Fasteners) & Continuous (Welds & bonding)
- Sketching SN curves & understanding their shape

2. Design Methods

- Safety Factor vs. SN Curve vs. Fracture Mechanics

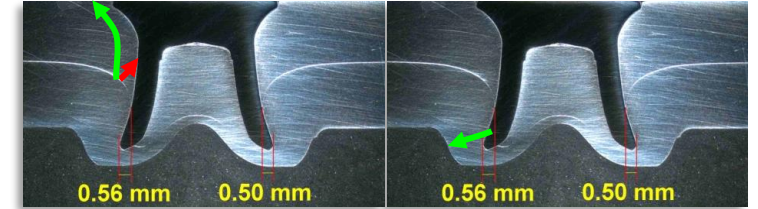
3. Fatigue Testing & Characterisation

- Measuring Load vs. Life curves
- Transforming Load vs. Life to 'Stress' vs. Life

4. In search of a Robust 'Stress' Parameter

- Local vs. Nominal vs. Hot-spot Stresses
- Stress (*Continuous*) vs. Load (*Discrete*) Recovery

5. Conclusion

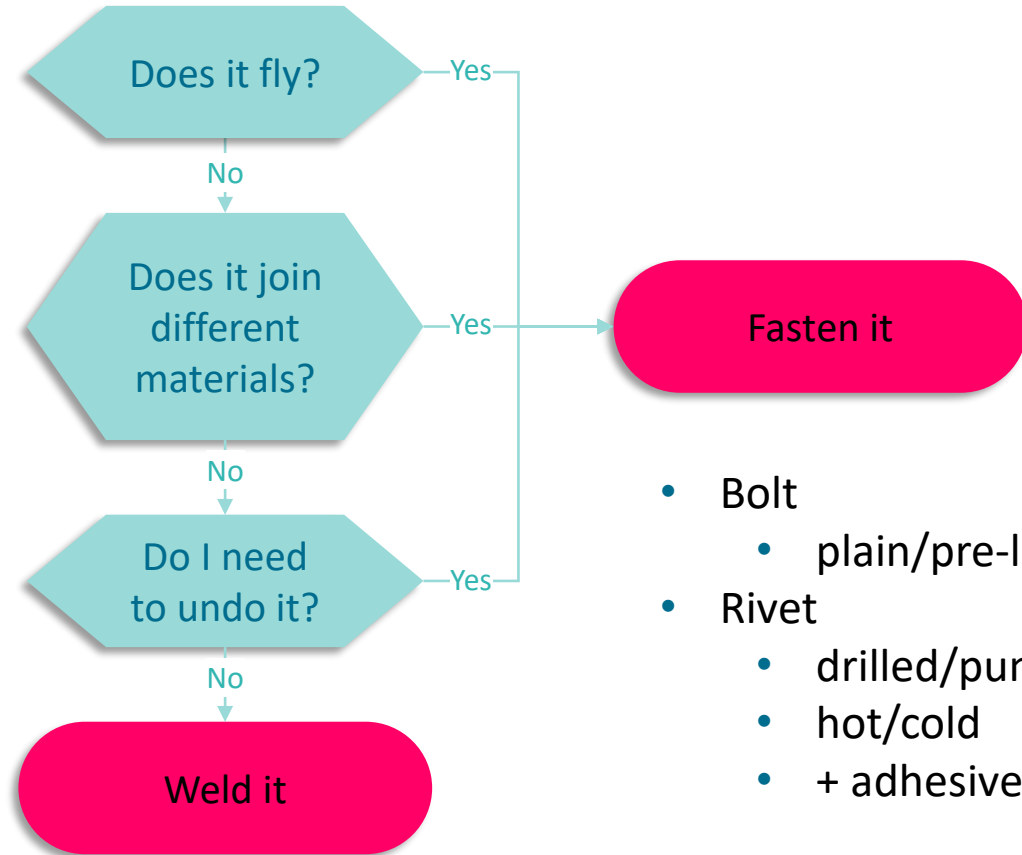
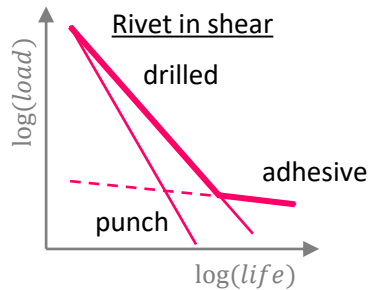
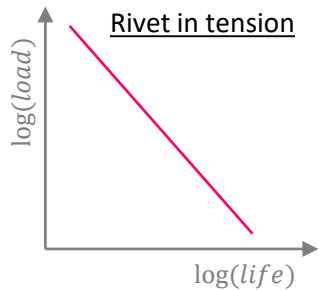
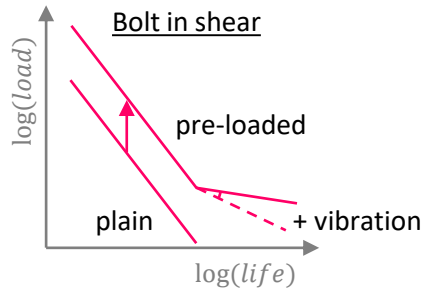
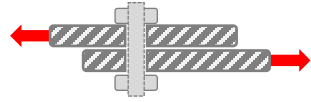
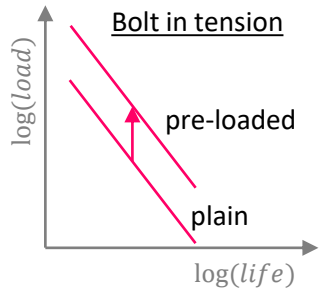
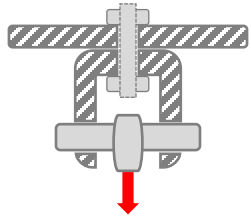


JLR research on SPRs



Hyundai RM15 Concept

The olden days



- Bolt
 - plain/pre-loaded
- Rivet
 - drilled/punch
 - hot/cold
 - + adhesive

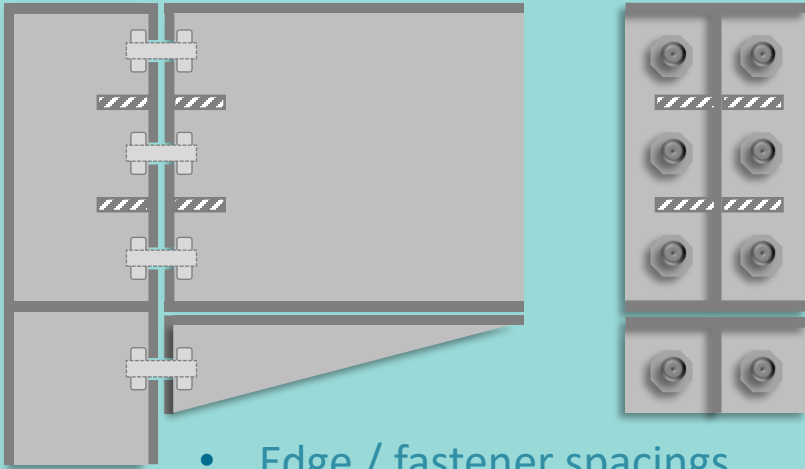
Shape of SN curve depends on the failure modes

- **Load type:** *axial vs. shear*
- **Secondary effects:** *mean stress & surface finish*
- **Additional load paths:** *additional friction*
- **Alternative failure modes:** *fastener/sheet (in shear or bearing)*

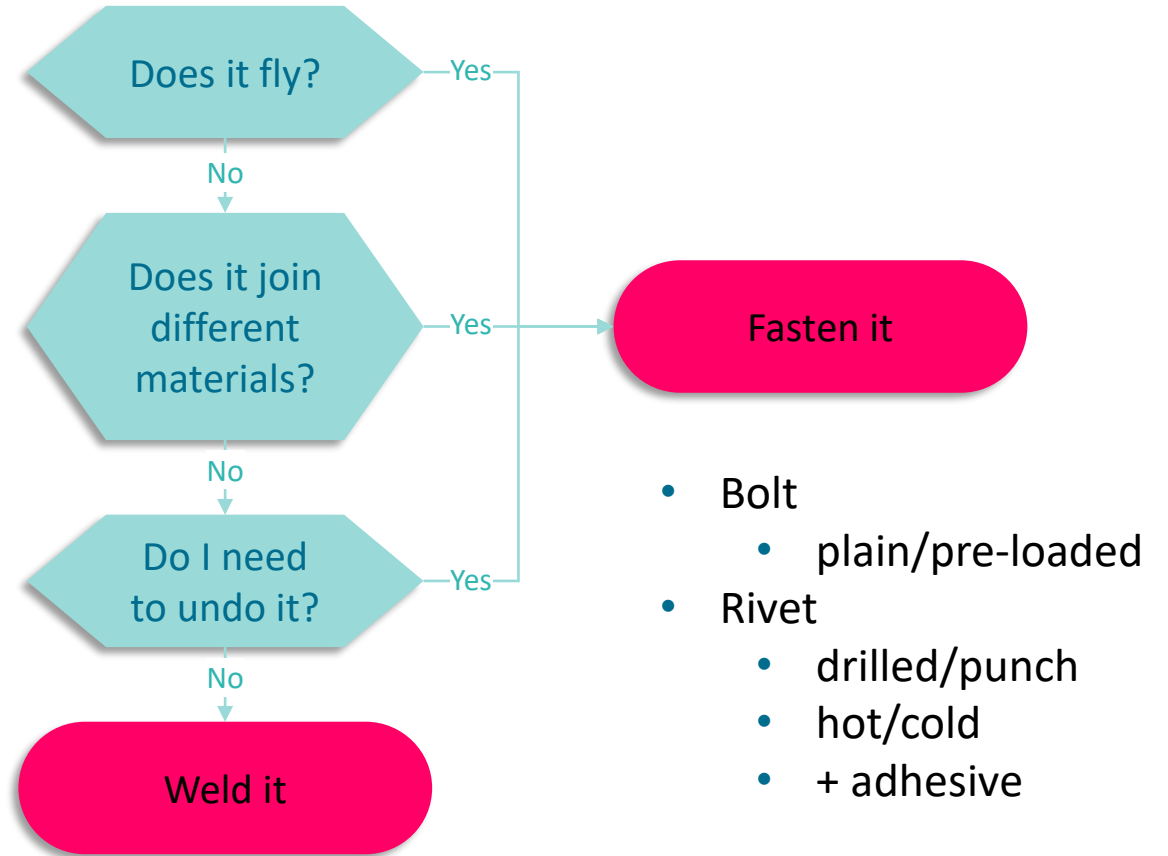
The olden days

Good practice:

Ensures a consistent failure mode by avoiding suboptimal failures



- Edge / fastener spacings
- Fastener vs. sheet thickness ratios
- Bearing stiffeners
- Butt straps
- etc...

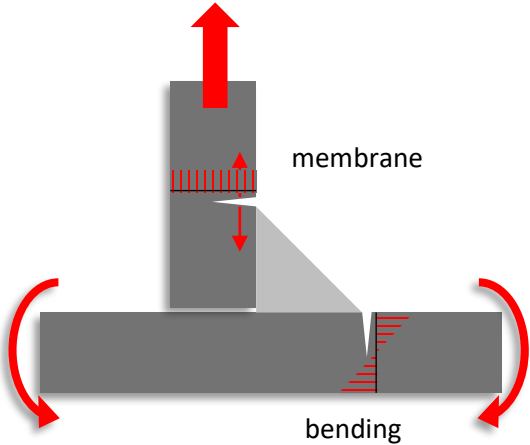


Shape of SN curve depends on the failure modes

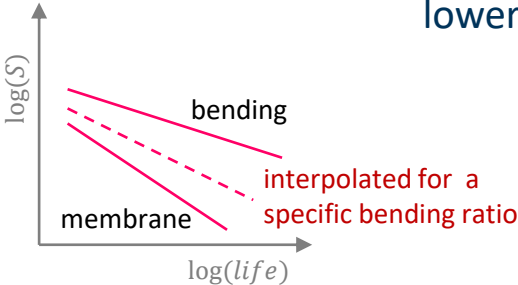
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The olden days

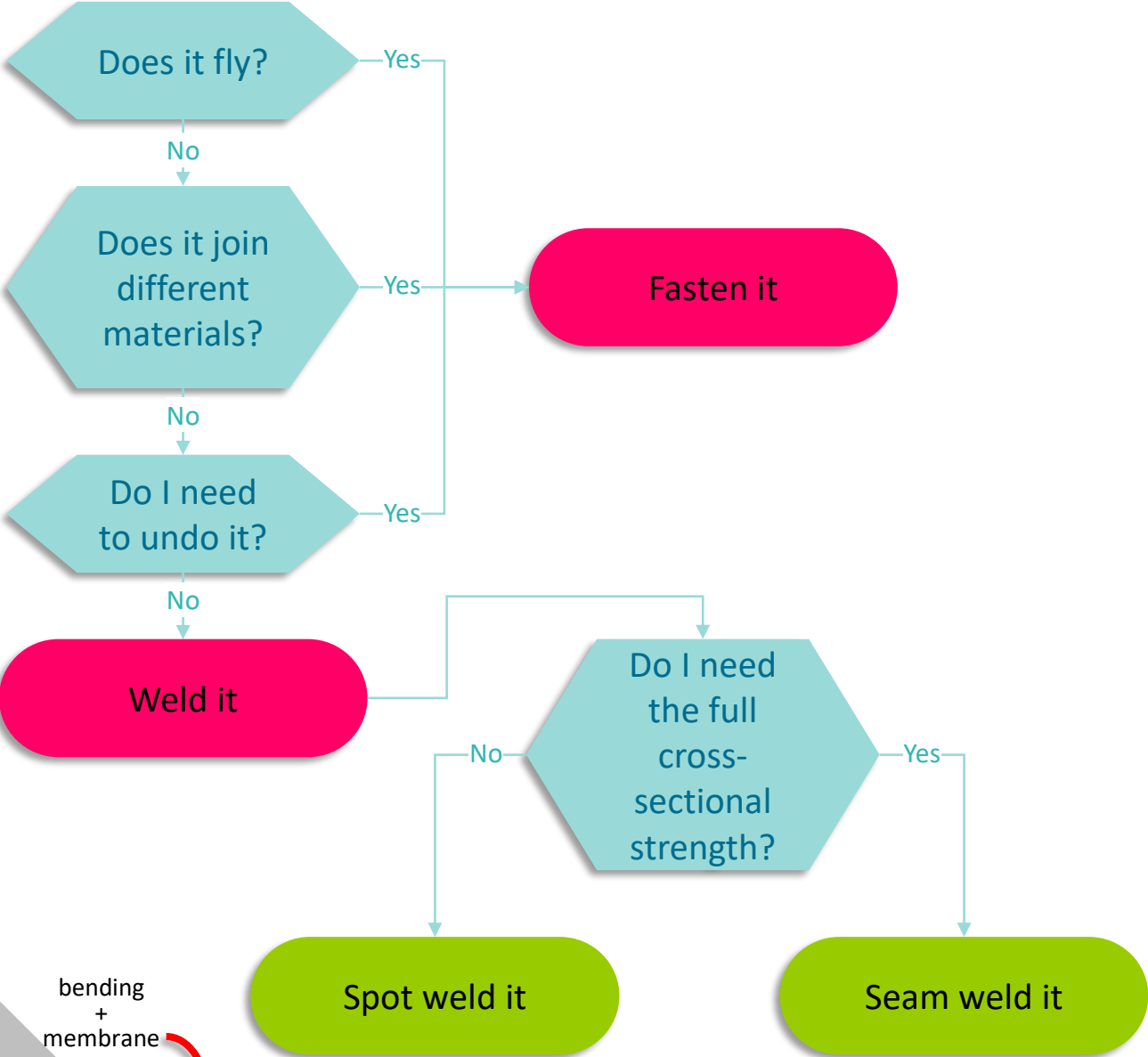
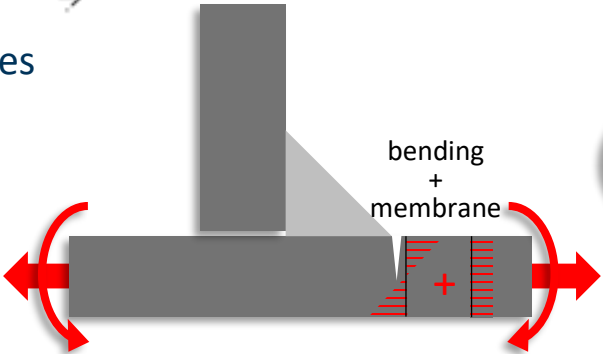
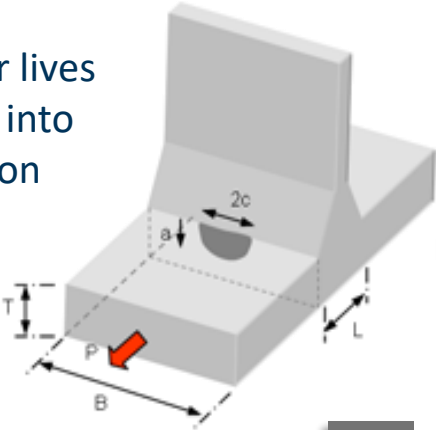
Seam weld



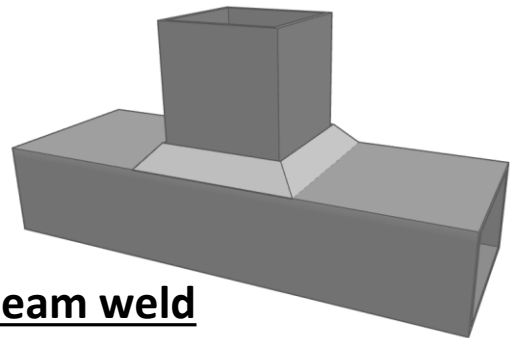
Bending = longer lives
∴ crack growing into lower stress region



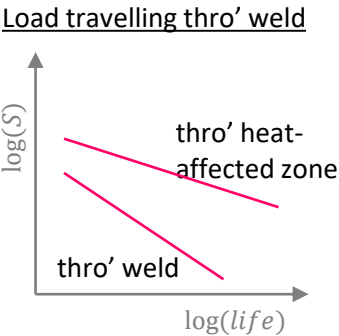
Membrane = shorter lives
∴ crack growing into uniform stress region



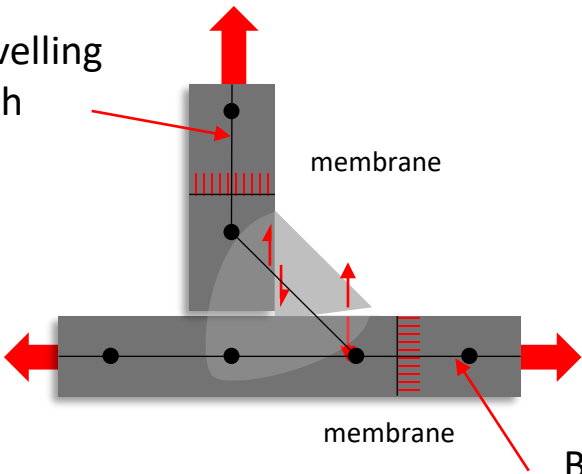
The olden days



Seam weld

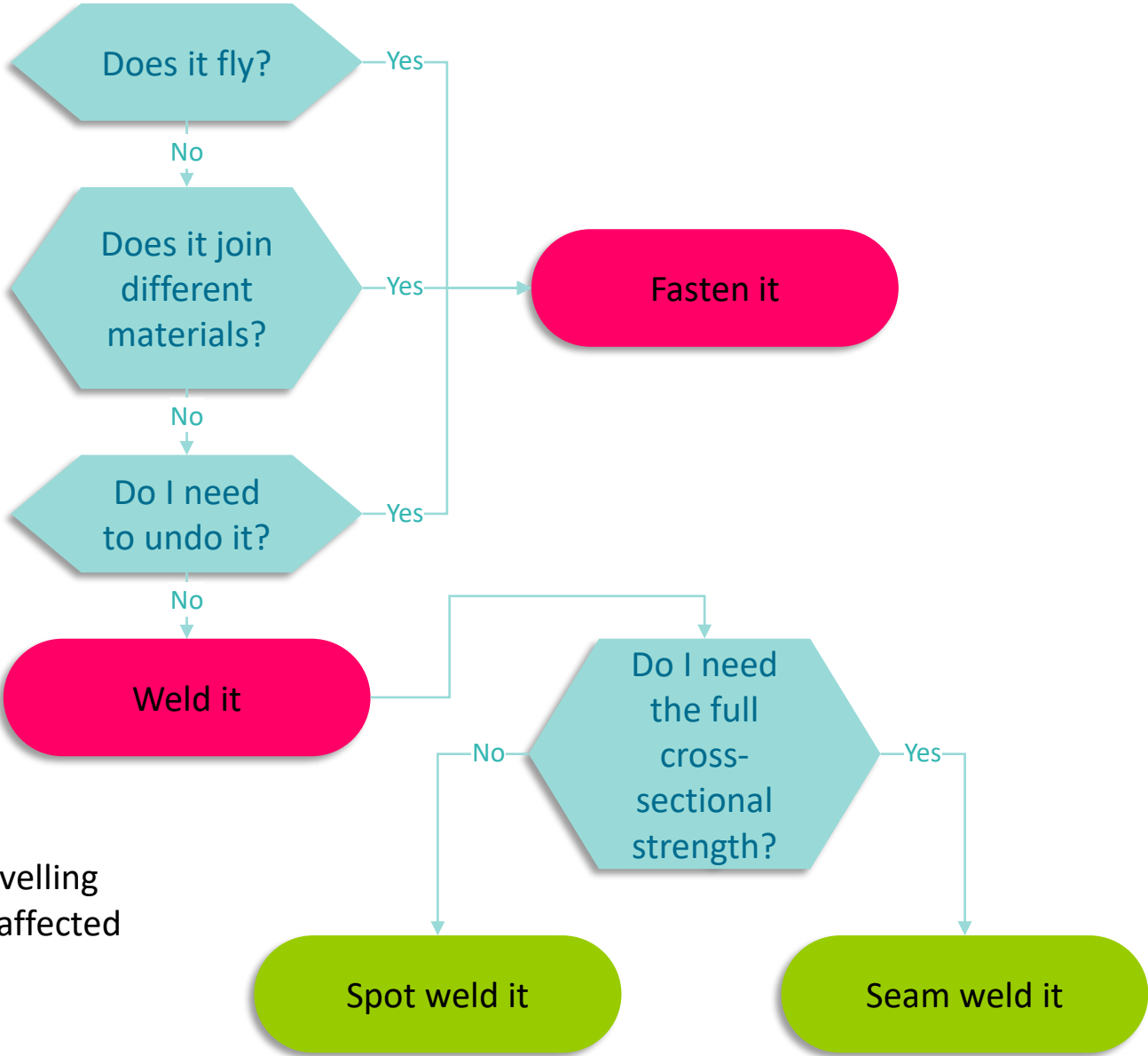


This load is travelling entirely through the weld



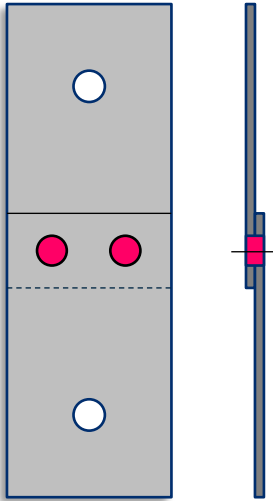
But this load is travelling through the heat-affected zone

We model this explicitly in FEA and capture the worst-case in the SN curve (load travelling entirely through the weld)

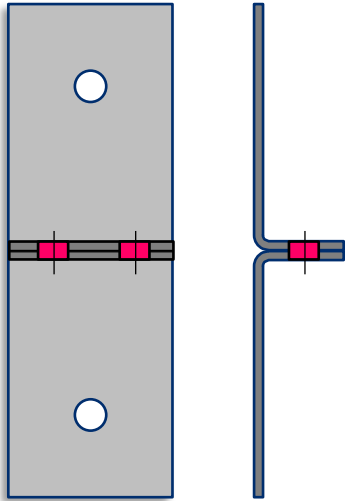


The olden days

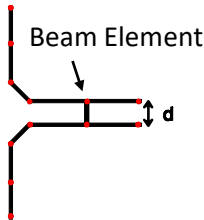
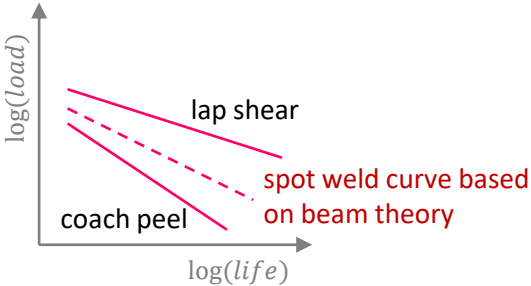
Spot weld



Lap shear joint

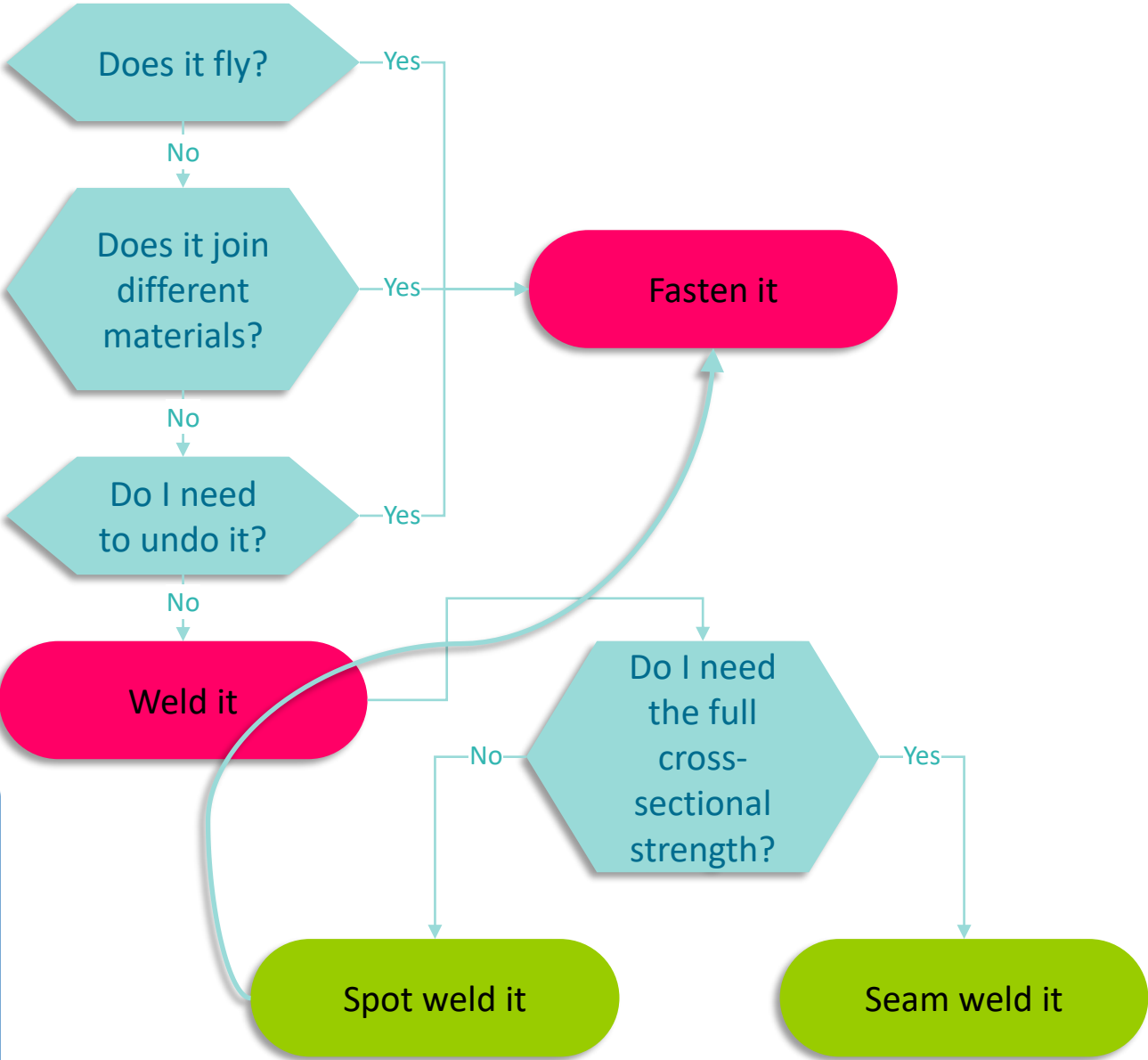


Coach peel joint

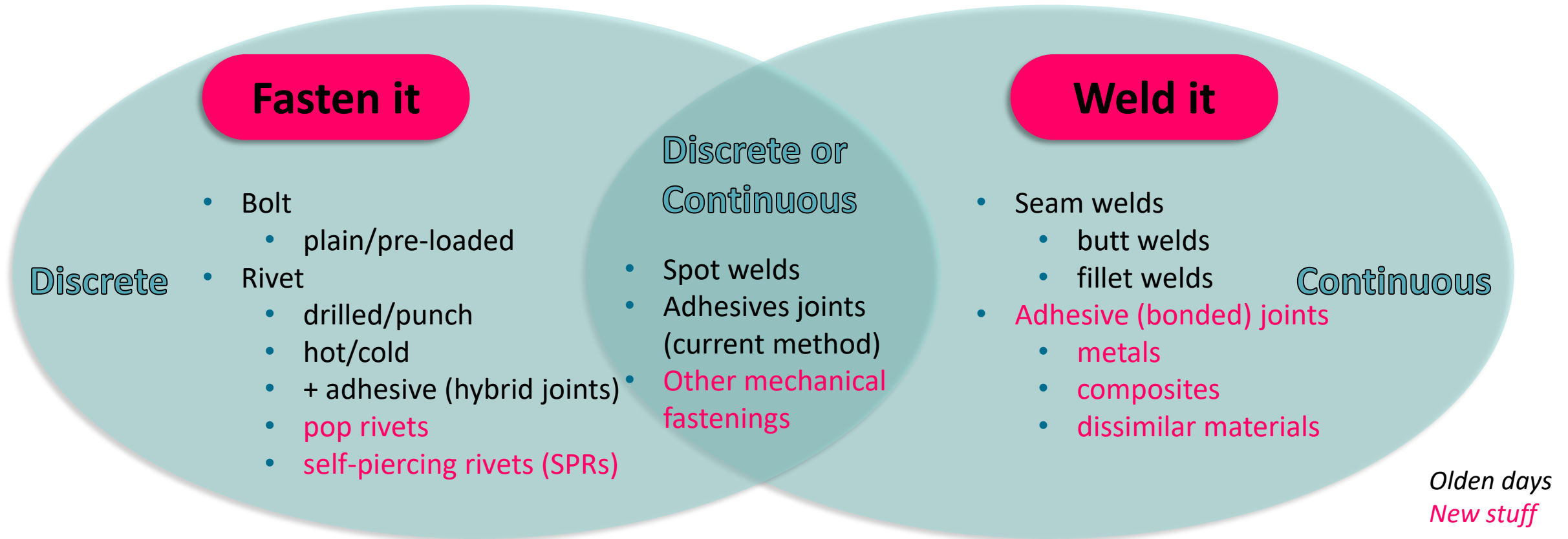


Equations to calculate structural stresses in the sheet

$$\Delta S(\theta) = -\sigma(F_x)\cos\theta - \sigma(F_y)\sin\theta + \sigma(F_x) + \sigma(M_x)\sin\theta + \sigma(M_y)\cos\theta$$
$$\sigma(F_x) = d^{DEFXY} \cdot t^{TEFXY} \cdot \frac{F_x}{\pi dt}$$
$$\sigma(F_y) = d^{DEFXY} \cdot t^{TEFXY} \cdot \frac{F_y}{\pi dt}$$
$$\sigma(F_z) = SFZ \cdot d^{DEFZ} \cdot t^{TEFZ} \cdot \frac{1.744 F_z}{t^2}$$
$$\sigma(M_x) = SFMXY \cdot d^{DEMXY} \cdot t^{TEMXY} \cdot \frac{1.872 M_x}{dt^2}$$
$$\sigma(M_y) = SFMXY \cdot d^{DEMXY} \cdot t^{TEMXY} \cdot \frac{1.872 M_y}{dt^2}$$



Summary so far – *Discrete* vs. *Continuous* Joints



Shape of SN curve depends on the failure modes

- **Load type:** *axial vs. shear or membrane vs. bending*
- **Secondary effects:** *mean stress & surface finish (+ thickness effects)*
- **Additional load paths:** *additional friction (hybrid joints)*
- **Alternative failure modes:** *fastener / sheet or sheet / nugget*

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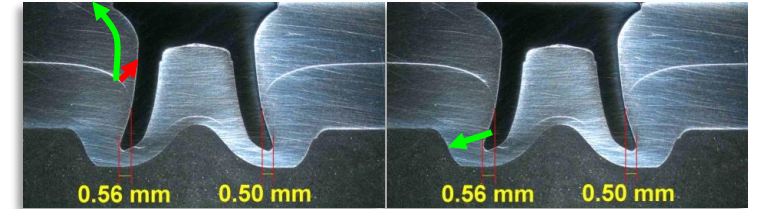
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- Stress (*Continuous*) vs. Load (*Discrete*) Recovery

5. Conclusion

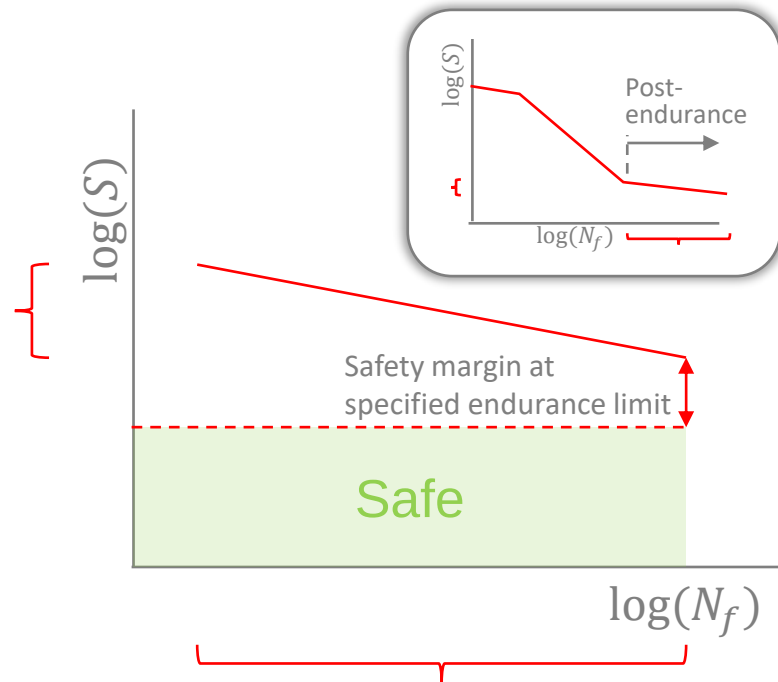


JLR research on SPRs



Hyundai RM15 Concept

Safety factor vs. SN vs. Crack growth modelling



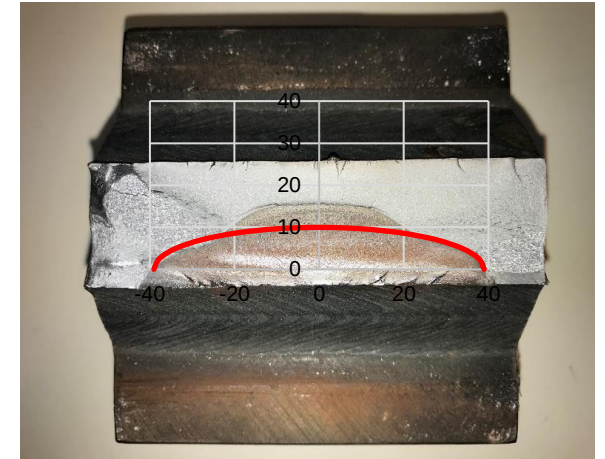
Safety factor approach

- Use when SN curve is flat
- Much better resolution in stress than life
- Ideal for traditional adhesive joints (brittle behaviour)
- Ideal if there's post-endurance behaviour (very high-cycle)



SN approach

- Use when SN curve is steep
- Good resolution in life and stress, but life is more intuitive for durability/reliability targets
- Ideal for joints in metals and composites
- Ideal for some modern adhesives (ductile behaviour)

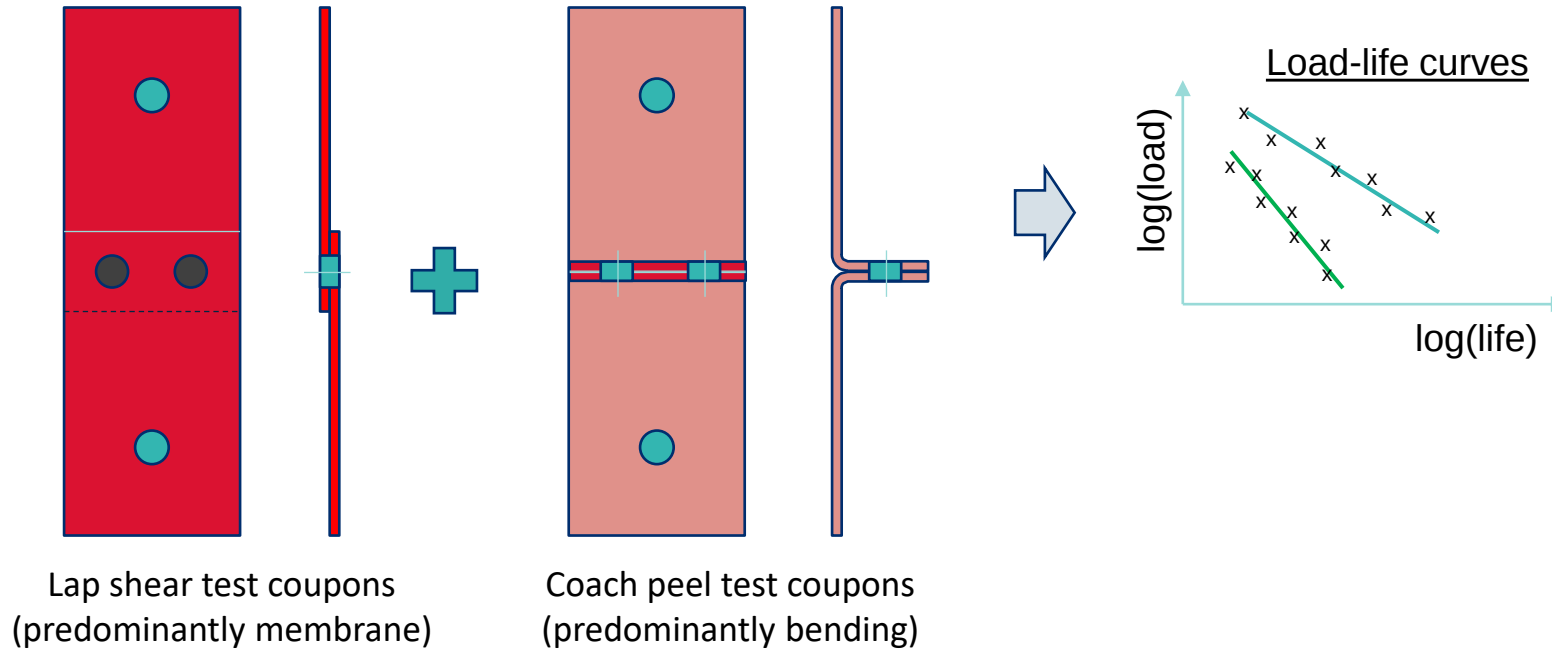


WholeLife

Fracture mechanics approach

- Use when there's significant crack growth through a complex stress field
- Life based on crack growth modelling is the most accurate
- Ideal for 'designed welds' (i.e. dressed, stress relieved, built-up)
- Ideal for damage tolerant structures

Fatigue testing and characterization (curve-fitting)



But we need 'Stress' – life curves!

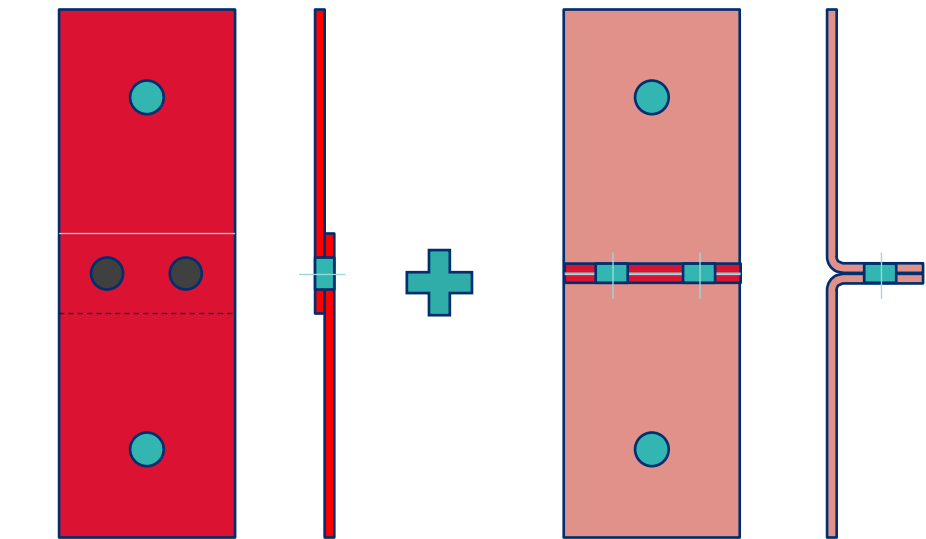
How to get them?

Specimen joints must be as close as possible to production

- Similar geometries (butt, fillet weld, etc)
- Similar penetration for welds
- Similar production quality
- Similar material thicknesses (*may derive an explicit factor*)
- etc.

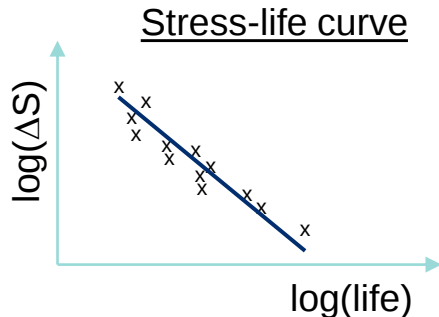
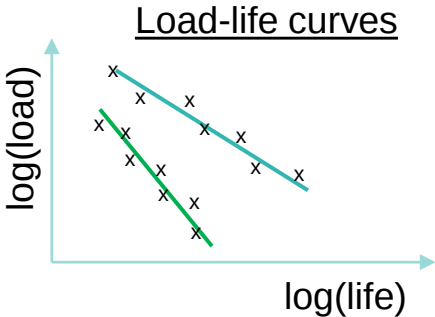
Fatigue testing and characterization (curve-fitting)

The final conundrum:
What is 'Stress'

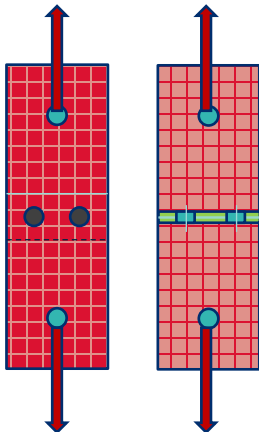


Lap shear test coupons
(predominantly membrane)

Coach peel test coupons
(predominantly bending)



Use the same FE
meshing, analysis
methods & stress
recovery methods
as the customer



FEA analysis

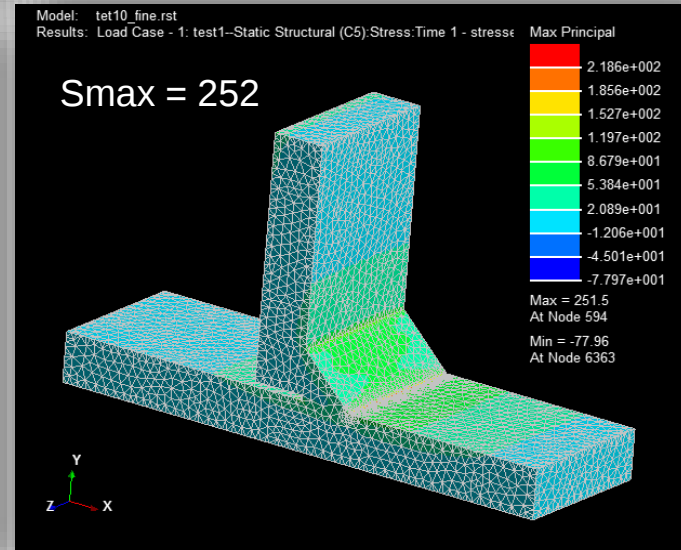
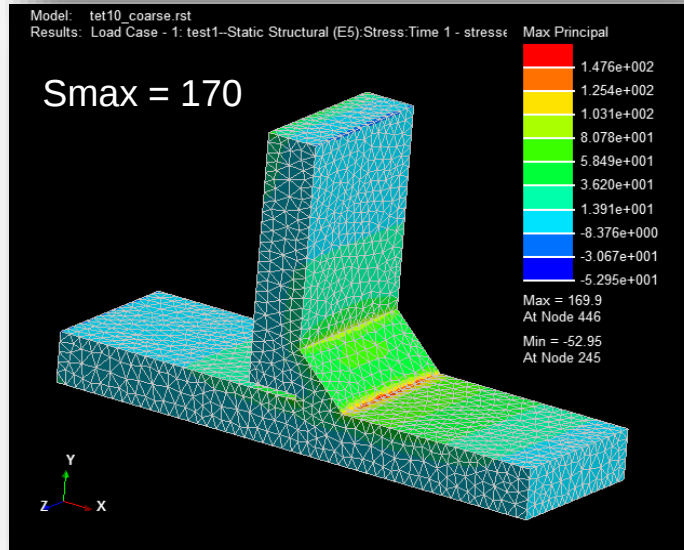
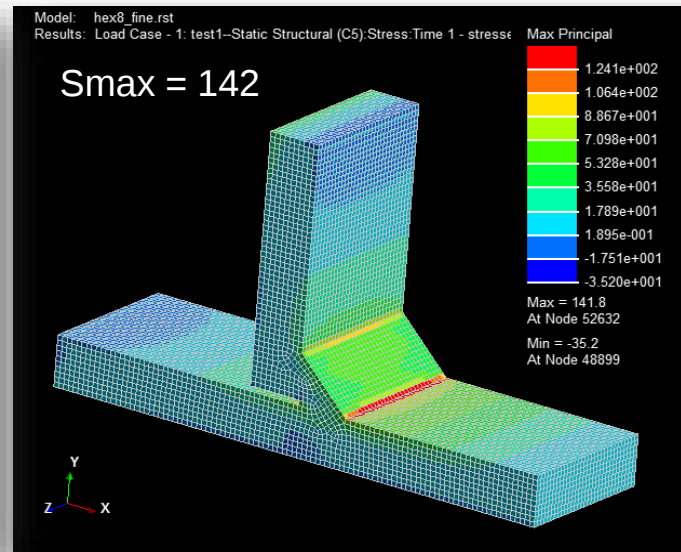
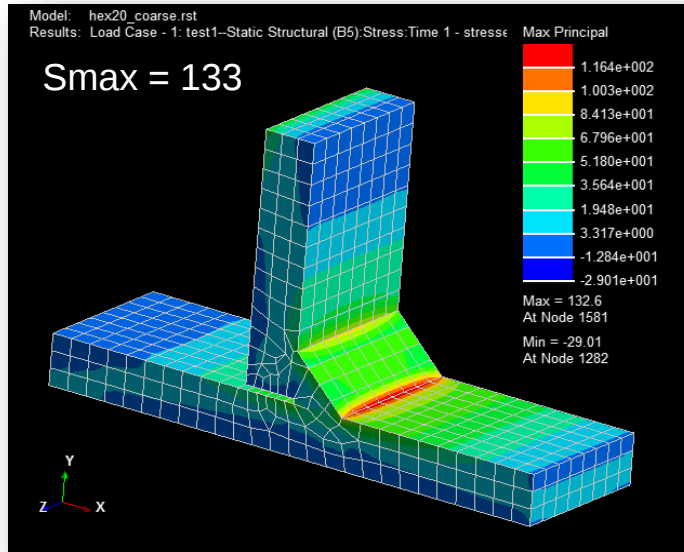
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Optimizer

Component	Factor	Diameter Exponent	Thickness Exponent
FX, FY	SFFXY = 1.0	DEFXY = 0.0	TEFXY = 0.0
MX, MY	SFMXY = 0.6	DEMXY = 0.0	TEMXY = 0.5
FZ	SFFZ = 0.6	DEFZ = 0.0	TEFZ = 0.5

Defining 'Stress' – *Local stress* vs. *Nominal stress* vs. *Hotspot stress*



Requirements for the 'Stress' parameter

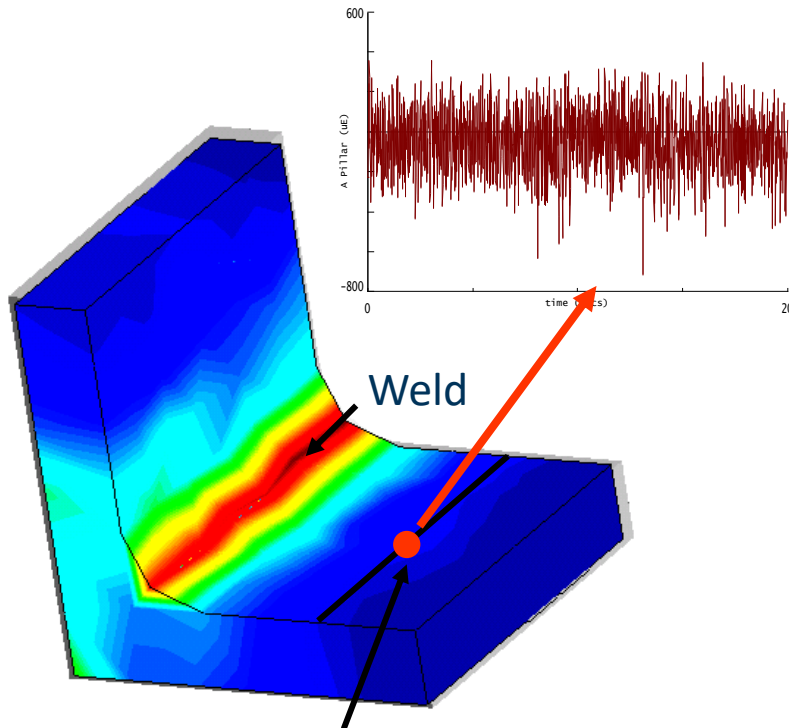
- Robust estimate of 'stress' consistent with:
 - avoiding mesh singularities
 - uncertainties in surface quality/finish
 - uncertainties in residual stresses
 - joint-joint statistical variability
 - natural scatter of fatigue test results
- 'Stress' parameter must be **consistent** between structure mesh and coupon mesh (i.e. same mesh guidelines)

"Consistent inaccuracy beats inconsistent accuracy!"

Examples showing how critical stress changes with mesh

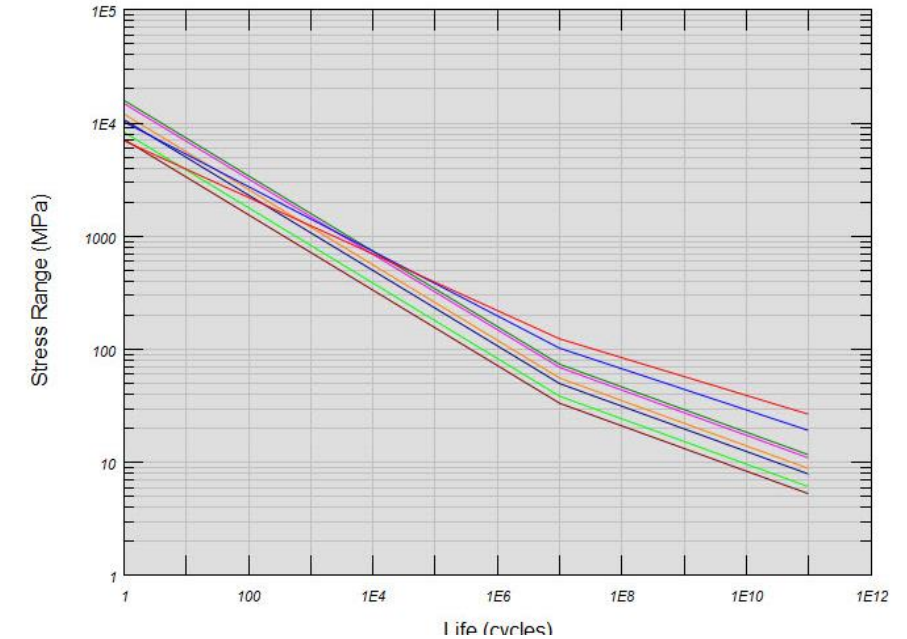
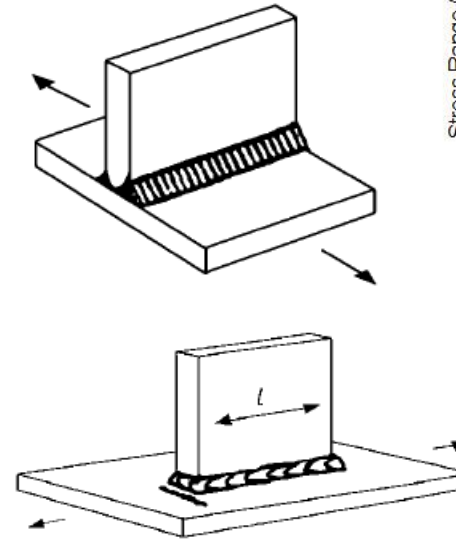
Defining 'Stress' – *Local stress* vs. *Nominal stress* vs. *Hotspot stress*

'Stress' definition



'Stress' vs. life relationship

- Maintain a consistent definition of 'stress'



Legend			
Steel Weld BS5400 ClassB::Stress-life (S-N)	SRI1: 6955 MPa	b1:-0.25 b2:-0.1667	NC1: 1e+007
Steel Weld BS5400 ClassC::Stress-life (S-N)	SRI1: 10222 MPa	b1:-0.2857 b2:-0.1818	NC1: 1e+007
Steel Weld BS5400 ClassD::Stress-life (S-N)	SRI1: 15861 MPa	b1:-0.3333 b2:-0.2	NC1: 1e+007
Steel Weld BS5400 ClassE::Stress-life (S-N)	SRI1: 14873 MPa	b1:-0.3333 b2:-0.2	NC1: 1e+007
Steel Weld BS5400 ClassF::Stress-life (S-N)	SRI1: 12005 MPa	b1:-0.3333 b2:-0.2	NC1: 1e+007
Steel Weld BS5400 ClassF2::Stress-life (S-N)	SRI1: 10714 MPa	b1:-0.3333 b2:-0.2	NC1: 1e+007
Steel Weld BS5400 ClassG::Stress-life (S-N)	SRI1: 8291 MPa	b1:-0.3333 b2:-0.2	NC1: 1e+007
Steel Weld BS5400 ClassW::Stress-life (S-N)	SRI1: 7179 MPa	b1:-0.3333 b2:-0.2	NC1: 1e+007

Position strain gauge at least 1 gauge length from root to reduce stress averaging

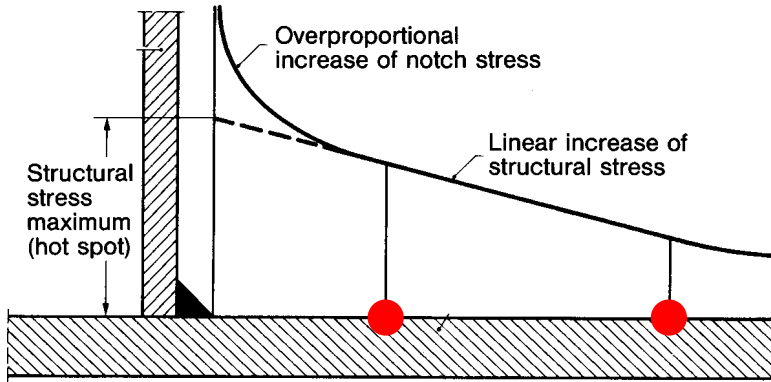
- Choosing a weld class or classes for the joint considering:

- Joint geometry (butt, fillet, etc.)
- Loading direction and mode
- Failure location to be considered
- Type of weld (full penetration etc.)

Very constraining!

Defining 'Stress' – Local stress vs. Nominal stress vs. Hotspot stress

'Stress' definition

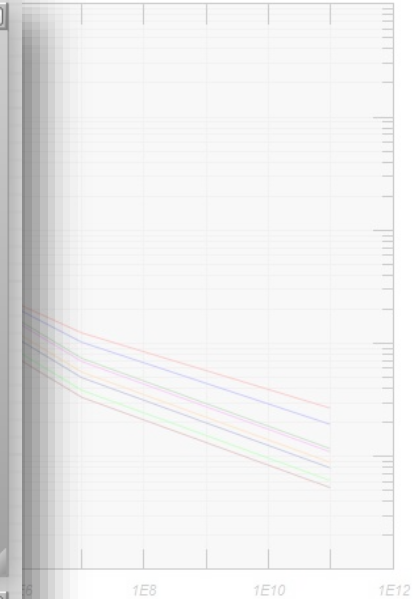
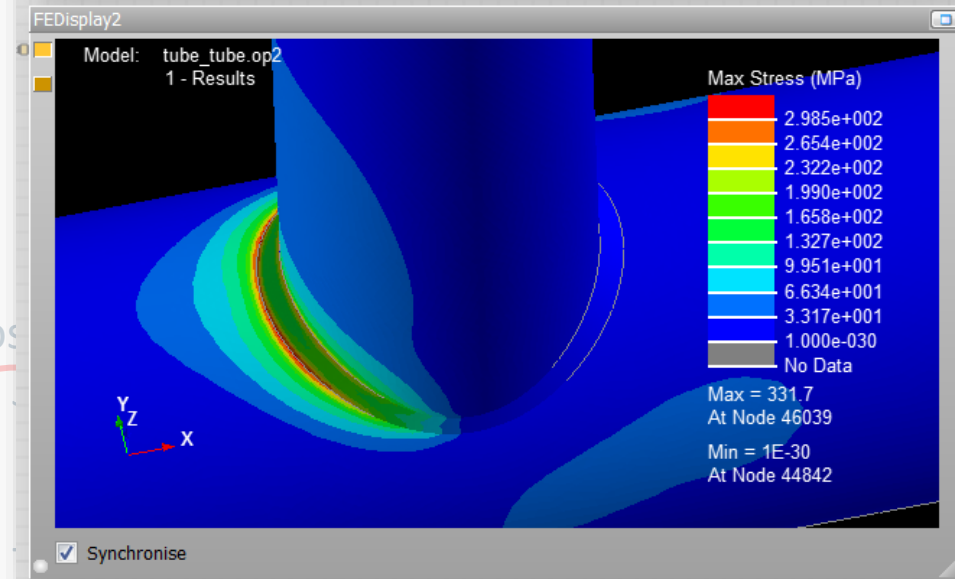
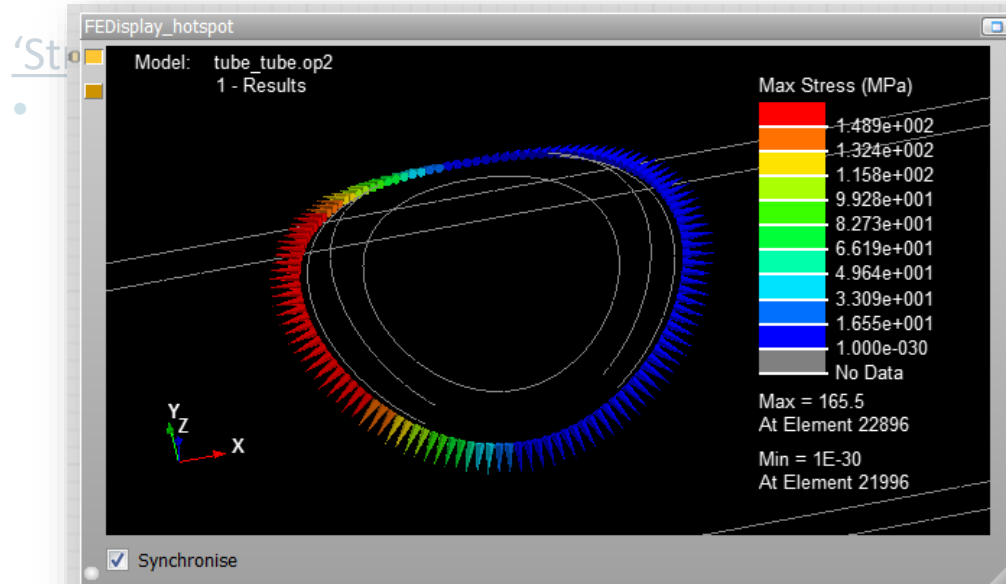


0.4 t
1.0 t

Surface Stress Extrapolation

'Surface Hotspot stress'

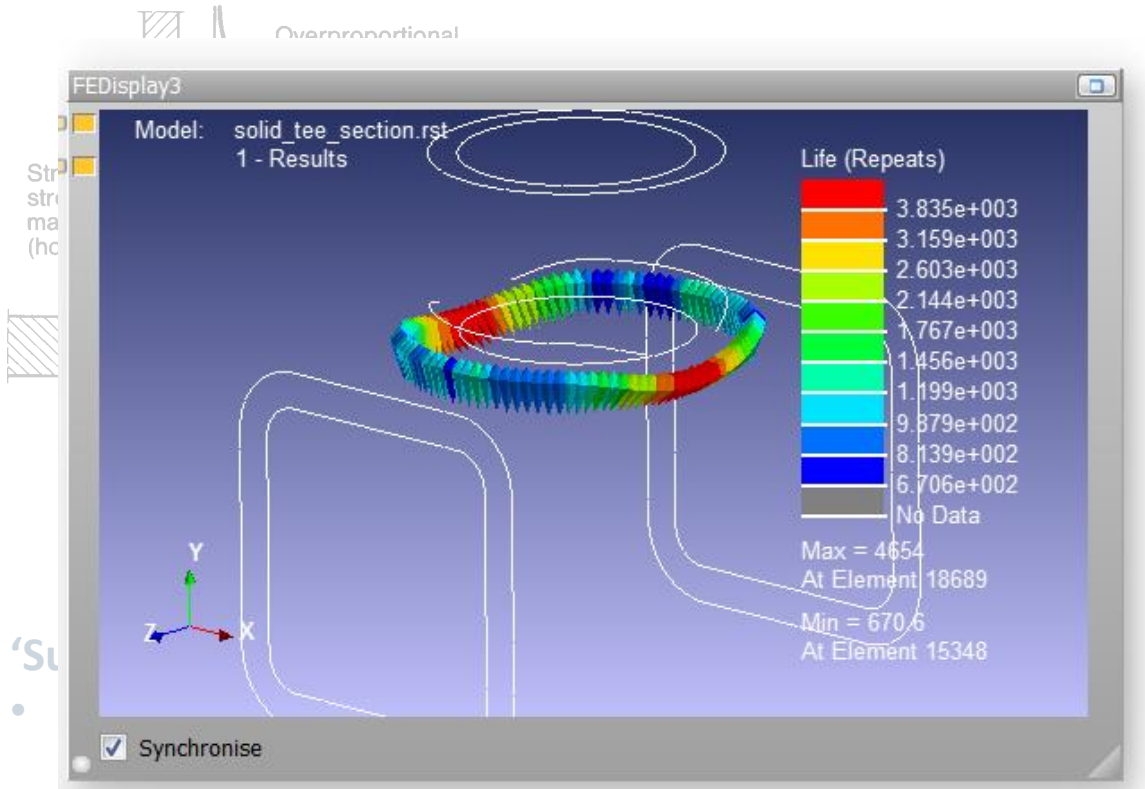
- mitigates constraints
 - ratio of thickness (IIW type a)
 - actual distance (IIW type b)
 - linear & parabolic extrapolations
- so DesignLife can be compatible with standard weld SN curves



R11: 6955 MPa	b1:-0.25	b2:-0.1667	NC1: 1e+007
R11: 10222 MPa	b1:-0.2857	b2:-0.1818	NC1: 1e+007
R11: 15861 MPa	b1:-0.3333	b2:-0.2	NC1: 1e+007
R11: 14873 MPa	b1:-0.3333	b2:-0.2	NC1: 1e+007
R11: 12005 MPa	b1:-0.3333	b2:-0.2	NC1: 1e+007
SR11: 10714 MPa	b1:-0.3333	b2:-0.2	NC1: 1e+007
R11: 8291 MPa	b1:-0.3333	b2:-0.2	NC1: 1e+007
SR11: 7179 MPa	b1:-0.3333	b2:-0.2	NC1: 1e+007

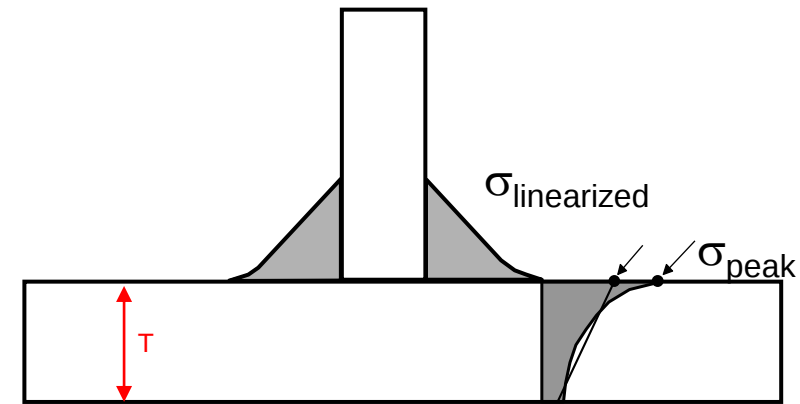
Defining 'Stress' – *Local stress* vs. *Nominal stress* vs. *Hotspot stress*

'Stress' definition



- actual distance (IIW type b)
- linear & parabolic extrapolations
- so DesignLife can be compatible with standard weld SN curves

Alternative 'Hotspot Stress' definition



Through-thickness Stress Extrapolation

'Through-thickness Hotspot stress'

- eliminates impact of mesh singularities

$$\sigma_{linearized} = \sigma_m + \sigma_b = \frac{P}{A} + \frac{MT}{2I}$$

- different to 'Surface Hotspot Stress' so needs different SN curve (consistency)

Defining 'Stress' – *through a fastener*

Spot welds & fasteners (rivets, SPRs, bolts, etc.)

Method 1 – continuous 'stress' recovery method

- As previous slides (e.g. treat spot weld as a small seam weld)

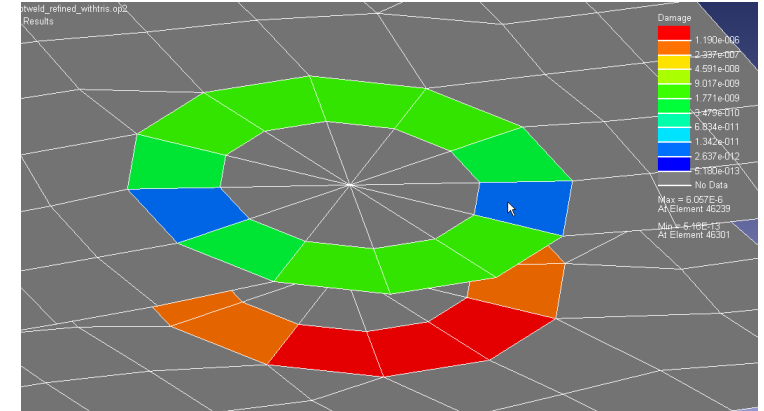
Method 2 – discrete load/displacement recovery method

Pros.

- Ideal for course mesh models (load transfer)
- load/displacement is converted to 'stress' using an engineering model external to FEA
- Reduced dependency on FEA package¹
- Many opportunities for stress model improvements (e.g. thickness corrections, multiple failure modes, critical plane analysis, etc.)

Cons.

- Multiple non-standard meshing methods
- Complexity of meshing (e.g. need for concentric meshes, etc.)
- Issues with over-stiffening of meshes
- Different methods on different FEA packages¹



Defining ‘Stress’ – *through a fastener*

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Method 1 – continuous ‘stress’ recovery method

- As previous slides (e.g. treat spot weld as a small seam weld)

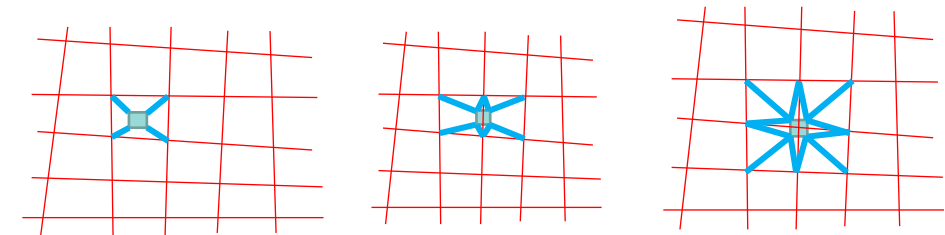
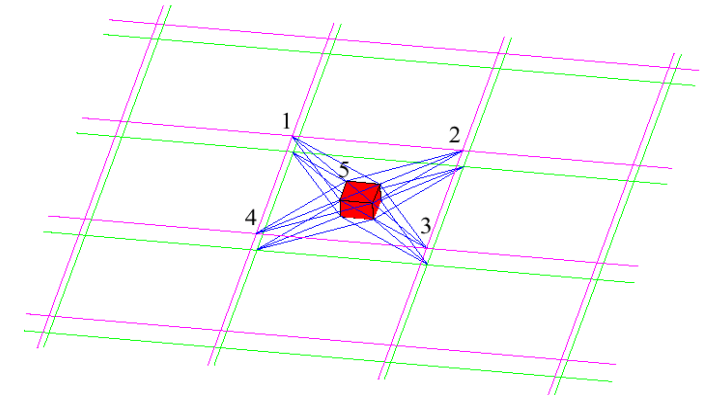
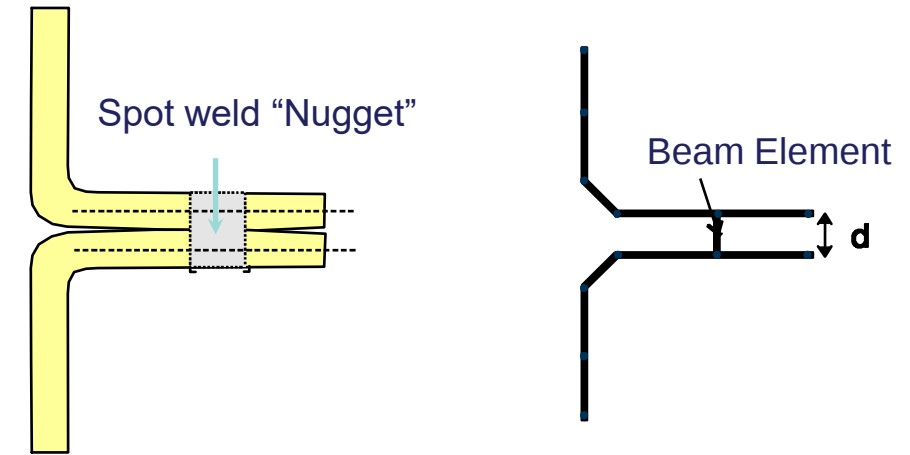
Method 2 – discrete load/displacement recovery method

Pros.

- Ideal for course mesh models (load transfer)
- load/displacement is converted to ‘stress’ using an engineering model external to FEA
- Reduced dependency on FEA package¹
- Many opportunities for stress model improvements (e.g. thickness corrections, multiple failure modes, critical plane analysis, etc.)

Cons.

- Multiple non-standard meshing methods
- Complexity of meshing (e.g. need for concentric meshes, etc.)
- Issues with over-stiffening of meshes
- Different methods on different FEA packages¹



Defining ‘Stress’ – *through a fastener*

Calculation of engineering stresses in a sheet based on forces $F_x, F_y, F_z, M_x, M_y, M_z$ obtained from the fastener reaction loads:

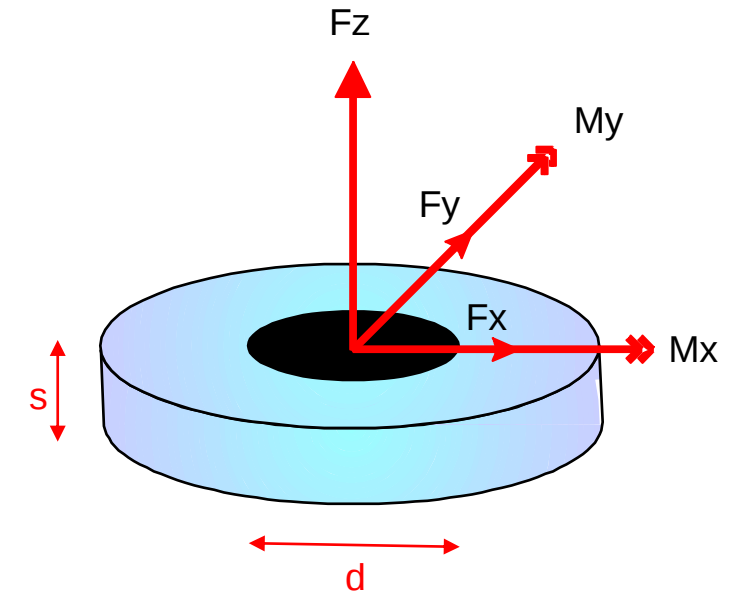
$$\begin{aligned} \sigma_{F_x} &= \left\{ \frac{F_{xy}}{\pi d s} (S_{F_{xy}}) (d^{D_{F_{xy}}}) (s^{T_{F_{xy}}}) \right. \\ \sigma_{F_y} &= \left. \frac{F_z}{s^2} (S_{F_z}) (d^{D_{F_z}}) (s^{T_{F_z}}) \right. \\ \sigma_{F_z} &= 1.744 \frac{F_z}{s^2} (S_{F_z}) (d^{D_{F_z}}) (s^{T_{F_z}}) \\ \sigma_{M_x} &= \left\{ 1.872 \frac{M_{xy}}{d s^2} (S_{M_{xy}}) (d^{D_{M_{xy}}}) (s^{T_{M_{xy}}}) \right. \\ \sigma_{M_y} &= \left. \frac{2}{\pi s d^2} \right. \\ \sigma_{M_z} &= \end{aligned}$$

Table 1: Transform coefficients used for steel sheets:

Component	Factor	Diameter exponent	Thickness exponent
F_x, F_y	$S_{F_{xy}} = 1.0$	$D_{F_{xy}} = 0$	$T_{F_{xy}} = 0$
M_x, M_y	$S_{M_{xy}} = 0.6$	$D_{M_{xy}} = 0$	$T_{M_{xy}} = 0.5$
F_z	$S_{F_z} = 0.6$	$D_{F_z} = 0$	$T_{F_z} = 0.5$

Table 2: Transform coefficients used for aluminium sheets:

Component	Factor	Diameter exponent	Thickness exponent
F_x, F_y	$S_{F_{xy}} = 0.4$	$D_{F_{xy}} = 0.5$	$T_{F_{xy}} = -0.25$
M_x, M_y	$S_{M_{xy}} = 0.4$	$D_{M_{xy}} = 0.5$	$T_{M_{xy}} = -0.25$
F_z	$S_{F_z} = 1.0$	$D_{F_z} = 0$	$T_{F_z} = 1.0$



Engineering ‘stress’ model for Spotweld fastener

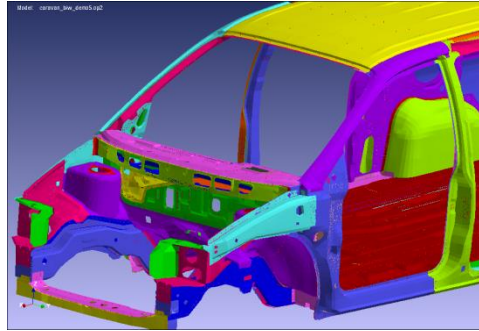
- Critical-plane calculation
- Correction factors available for relative scale, diameter and thickness
- Additional calculation for ‘nugget’ failure
- Additional calculation for 3-sheet joints

This versatile model has also been used for other fasteners (e.g. bolts, rivets and SPRs)
But DesignLife also allows user-defined models

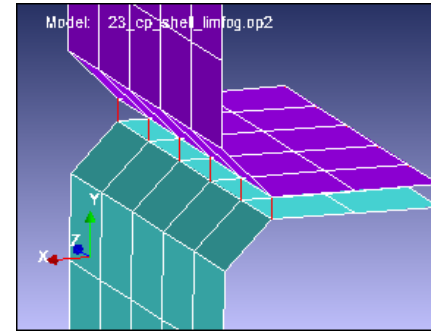
Defining 'Stress' – through an adhesive bond – current approach

This method recovers edge loads from an adhesive bond using beam elements and calculates the J integral for use as a safety factor analysis against J_{1c}

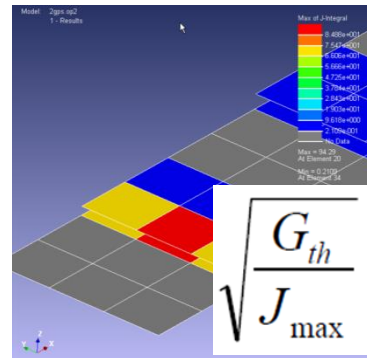
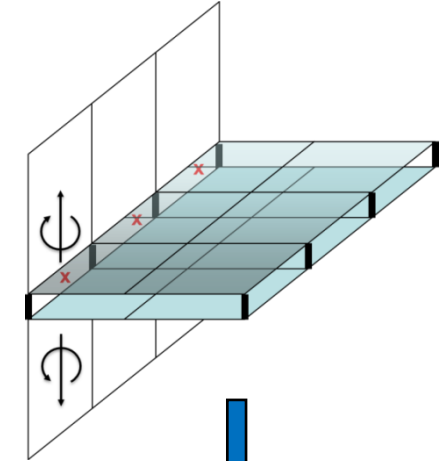
Global model



Simple joint modelling



Line forces and moments

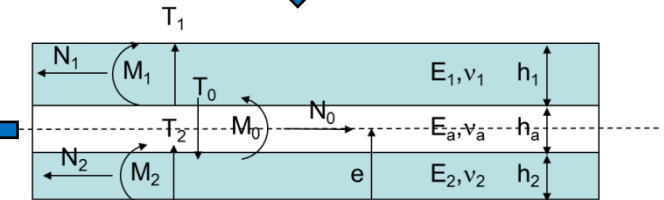


Postprocess Jmax and safety factor results

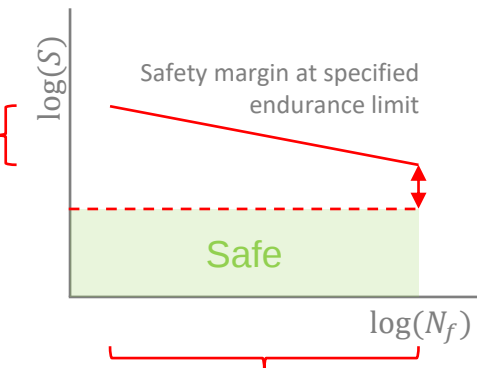
$$J_i = \frac{N_i^2}{2E_i h_i} + \frac{1.2 \cdot T_i^2}{2G_i h_i} + \frac{M_i^2}{2Q_i H_i} + T_i \phi_i$$

$$J_{tot} = J_1 + J_2 - J_0$$

Calculate J-integral history



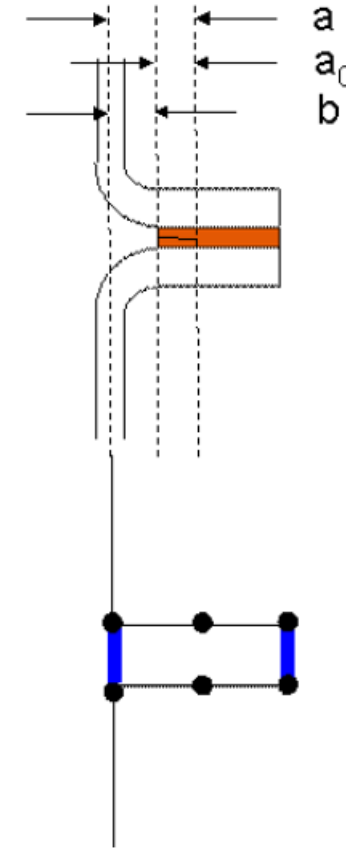
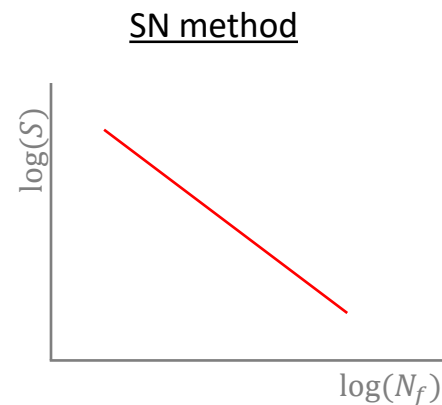
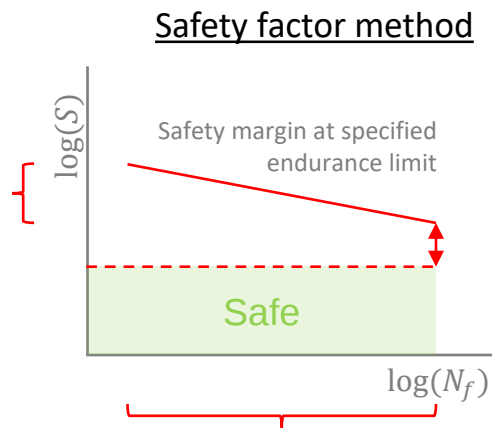
Transfer line forces to analytical sandwich model



Defining 'Stress' – *through an adhesive bond* – future approach

Future work:

- FEA packages now offer adhesive elements so the need for beam elements (& concentric meshes) is removed
- Many newer adhesives behave in a ductile manner so an SN method is preferred over the safety factor
- Adhesive bonds fail through fracture $\therefore$ research fracture mechanics approach



Agenda

1. Joint Classification

- Discrete (Fasteners) & Continuous (Welds & bonding)
- Sketching SN curves & understanding their shape

2. Design Methods

- Safety Factor vs. SN Curve vs. Fracture Mechanics

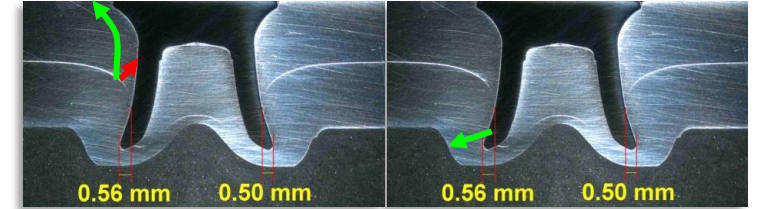
3. Fatigue Testing & Characterisation

- Measuring Load vs. Life curves
- Transforming Load vs. Life to 'Stress' vs. Life

4. In search of a Robust 'Stress' Parameter

- Local vs. Nominal vs. Hot-spot Stresses
- Stress (*Continuous*) vs. Load (*Discrete*) Recovery

5. Conclusion

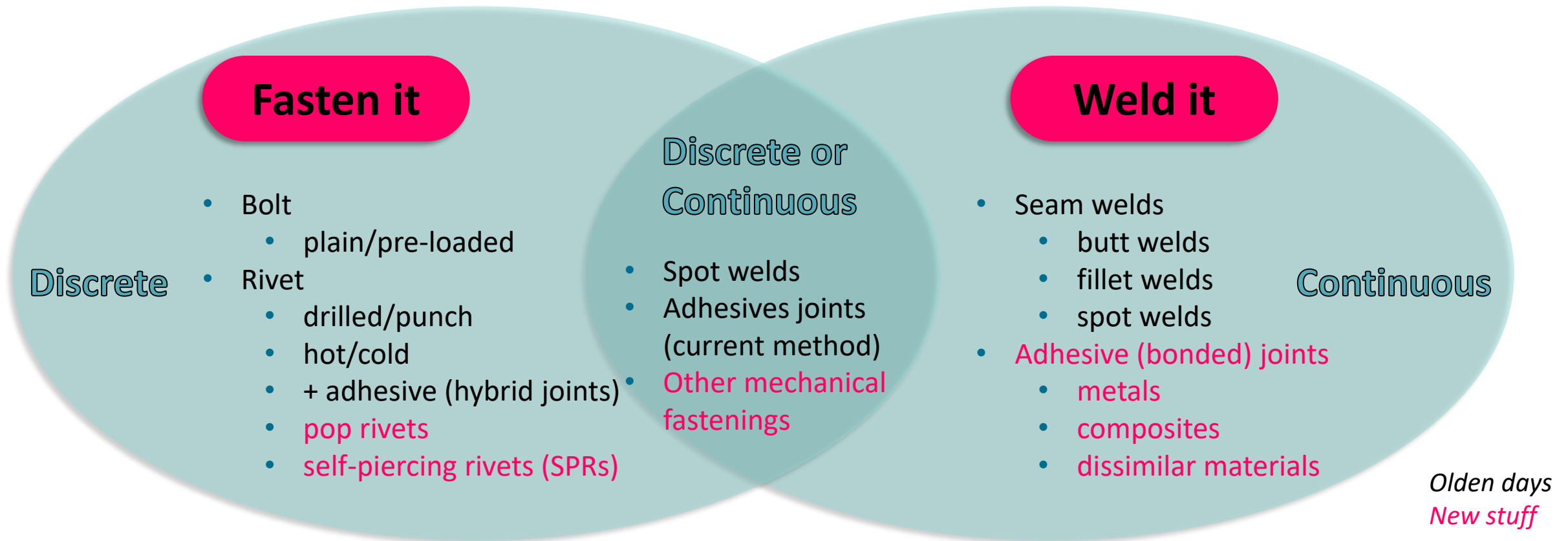


JLR research on SPRs



Hyundai RM15 Concept

Conclusion – *Discrete* vs. *Continuous Joints*



‘Stress’ parameter (**load/displacement recovery**)

- Load-transfer mesh
- Convert ‘Load’ to ‘engineering stress’
- Versatile method already included in DesignLife, but also user-programmable

‘Stress’ parameter (**mesh insensitive stress recovery**)

- Local vs. Nominal vs. Hot spot stresses (surface / through-thickness)
- Same mesh guidelines / analysis methods for fatigue curve fitting as in component design

“Consistent inaccuracy beats inconsistent accuracy!”

Q & A

Thank You

Dr Andrew Halfpenny | Director of Technology – nCode Products

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Q&A

▲ (no questions)

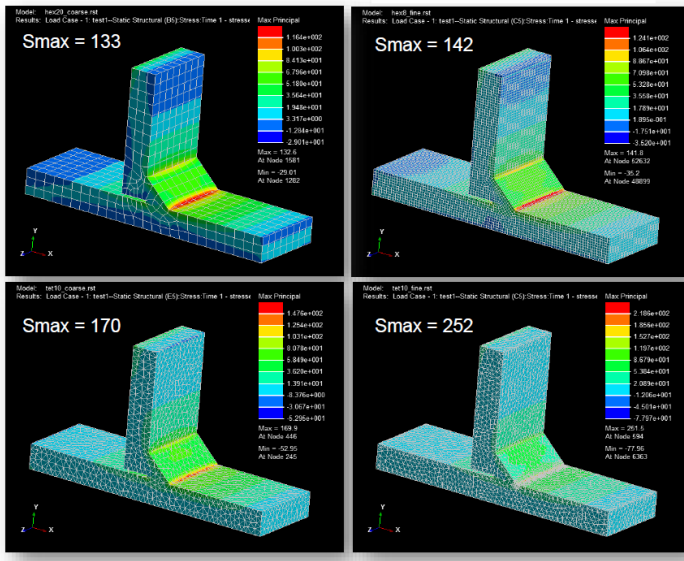


Prencia Technology Days

Virtual Seminar Series – Tuesday 2, 9, 16 November

Defining 'Stress'

"Consistent inaccuracy beats inconsistent accuracy!"



Introducing Fatigue Testing and Fatigue Simulation of Structural Joints

Presented by Dr. Andrew Halfpenny
Director of Technology
Prencia (HBK)

Tuesday, November 2, 2021