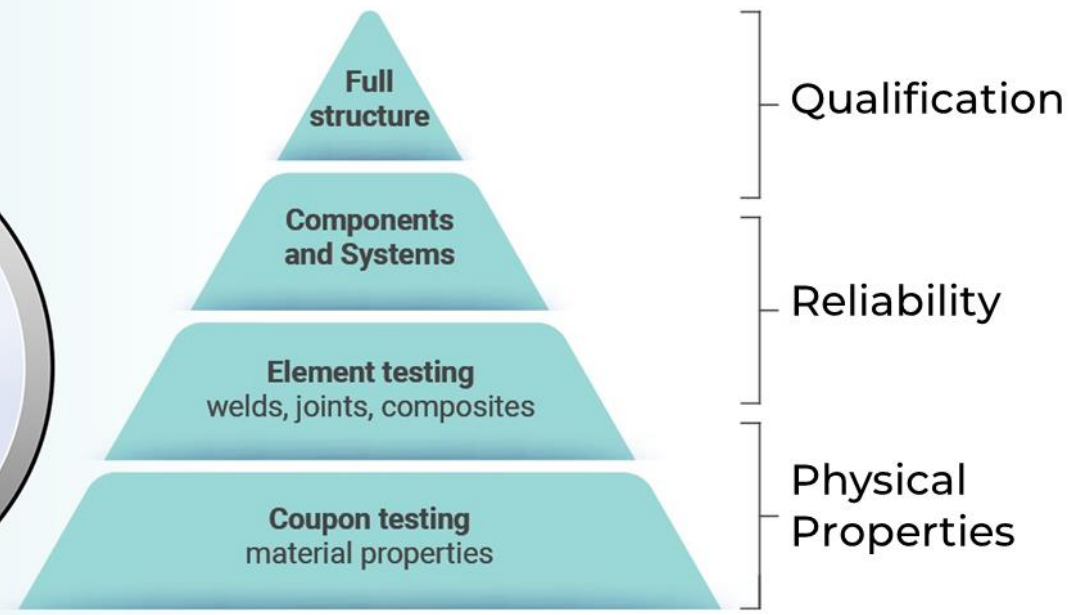
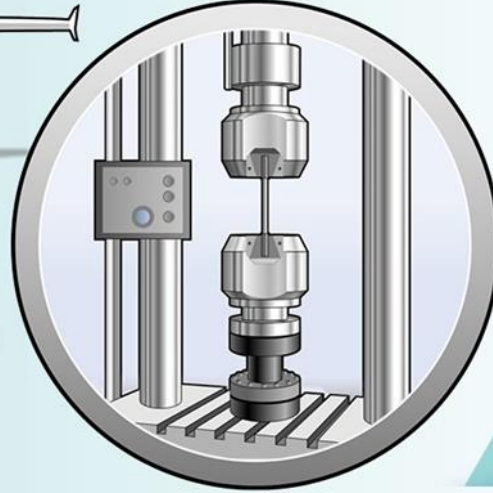
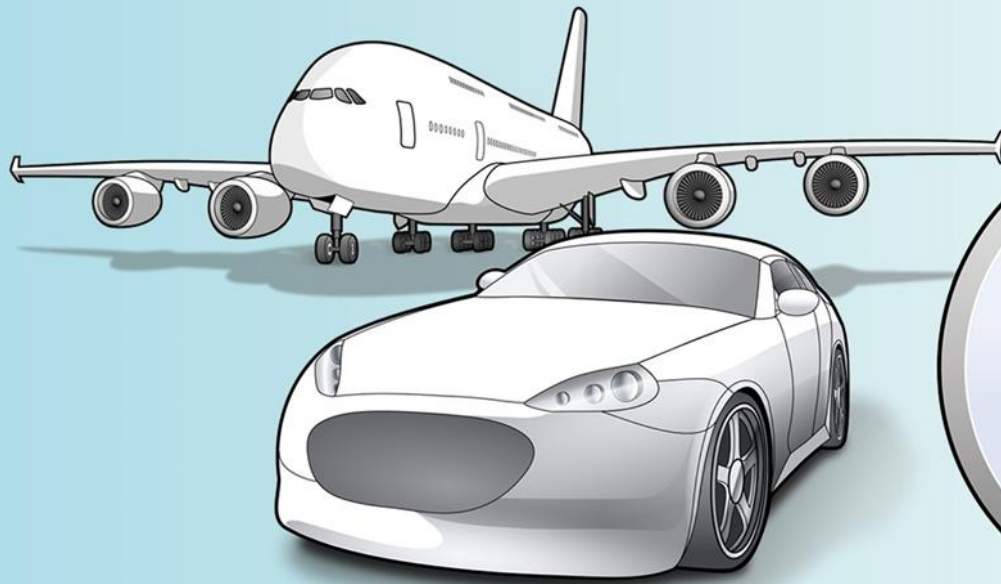
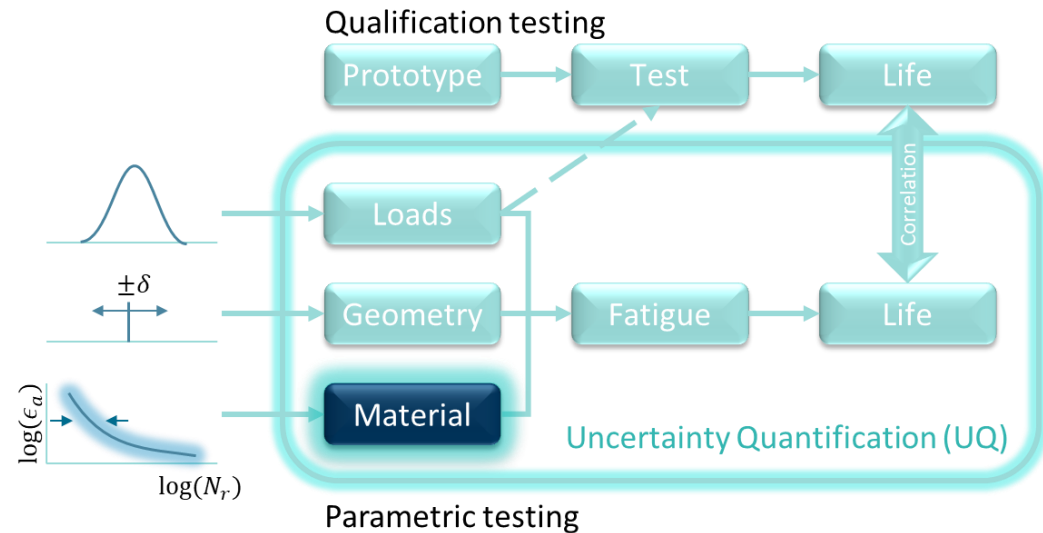




2023 HBK Technology Days

14 November 2023



- Complementary material parametric testing and simulation for fatigue prediction

Dr. Andrew Halfpenny,
Director of Technology – nCode Products, HBK

Complementary material parametric testing and simulation for fatigue prediction

Abstract

- The structural design of mechanical systems is significantly enhanced by simulation. Fatigue simulation is conveniently divided into two stages:
 1. Stress simulation (or constitutive modelling) is the transformation of loads, through displacements, to structural stresses. A structure may respond to these loads in either a static or dynamic fashion, and the response may be linear or non-linear in its nature. The Finite Element (FE) method is most commonly used for these simulations.
 2. Materials simulation is the transformation of these structural stresses into a statistical likelihood of failure. There are often many possible failure modes, and these may be classified as either:
 - Ultimate Limit State (ULS): e.g. tensile failure or buckling.
 - Serviceability Limit State (SLS): e.g. material ageing or fatigue.
- This presentation addresses the characterisation of increasingly complex materials. It considers the simple strength and elastic modulus values, and extends this to include the non-linear, elastic-plastic behaviour. Parameters are shown for several mathematical models depending on the complexity of the stress analysis required.
- The presentation continues by covering ULS and SLS failure models. These range from traditional static failures to SN and EN fatigue models, and on to complex thermo-mechanical fatigue and creep models.

Complementary material parametric testing and simulation for fatigue prediction

Abstract (continued)

- For one or more materials connected as a “joint” (bolts, rivets, welds, adhesives, etc.), mechanical testing alone is insufficient in determining their fatigue properties. In this case the term 'stress' will depend on the FE modelling strategy. Different strategies may result in different stress values. It is therefore essential that the test coupons are modelled using the same FE modelling strategy as the final components. This analysis is described in the presentation.
- Finally, it becomes necessary to consider the variability of material strength and how this affects the component design. The paper concludes by considering design parameters that are suitable for both traditional Deterministic designs (i.e. safe design curves), and parameters that are suitable for Stochastic design (i.e. simulating uncertainty).

About the presenter, Dr. Andrew Halfpenny, Director of Technology – nCode Products, HBK

- Dr. Halfpenny has a PhD in Mechanical Engineering from University College London (UCL) and a Master's in Civil and Structural Engineering. With over 25 years of experience in structural dynamics, vibration, fatigue and fracture, he has introduced many new technologies to the industry including: FE-based vibration fatigue analysis, crack growth simulation and accelerated vibration testing. He holds a European patent for the 'Damage monitoring tag' and developed the new vibration standard used for qualifying UK military helicopters.
- He has worked in consultancy with customers across the UK, Europe, Americas and the Far East, and has written publications on Fatigue, Digital Signal Processing and Structural Health Monitoring. He sits on the NAFEMS committee for Dynamic Testing and is a guest lecturer on structural dynamics with The University of Sheffield.

Complementary material parametric testing and simulation for fatigue prediction

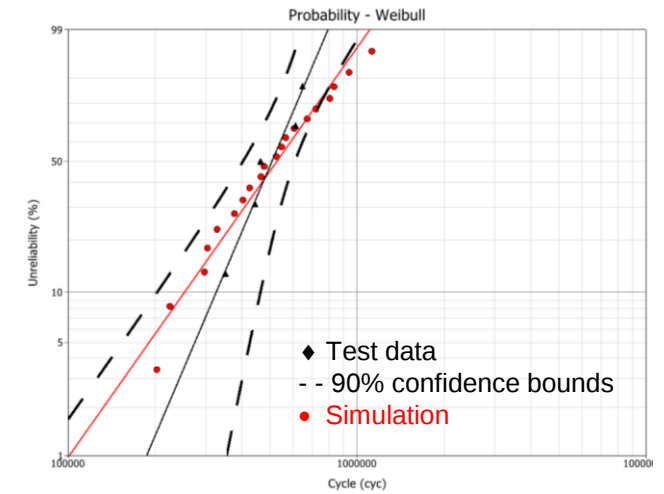
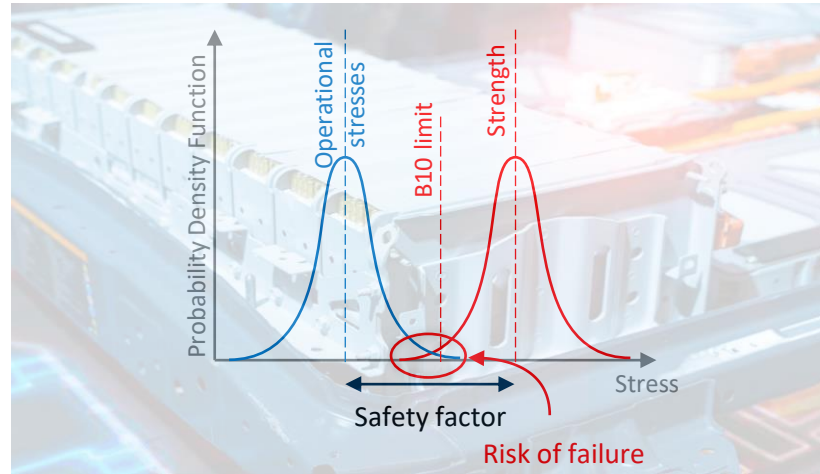
Presenter: Dr Andrew Halfpenny
Director of Technology – nCode Products



Motivation

Requirement:

For a **duty cycle** of 10 years / 150,000 miles,
I require **reliability** exceeding 95%
(i.e. fewer than 5% failures),
with a **confidence level** of at least 90%

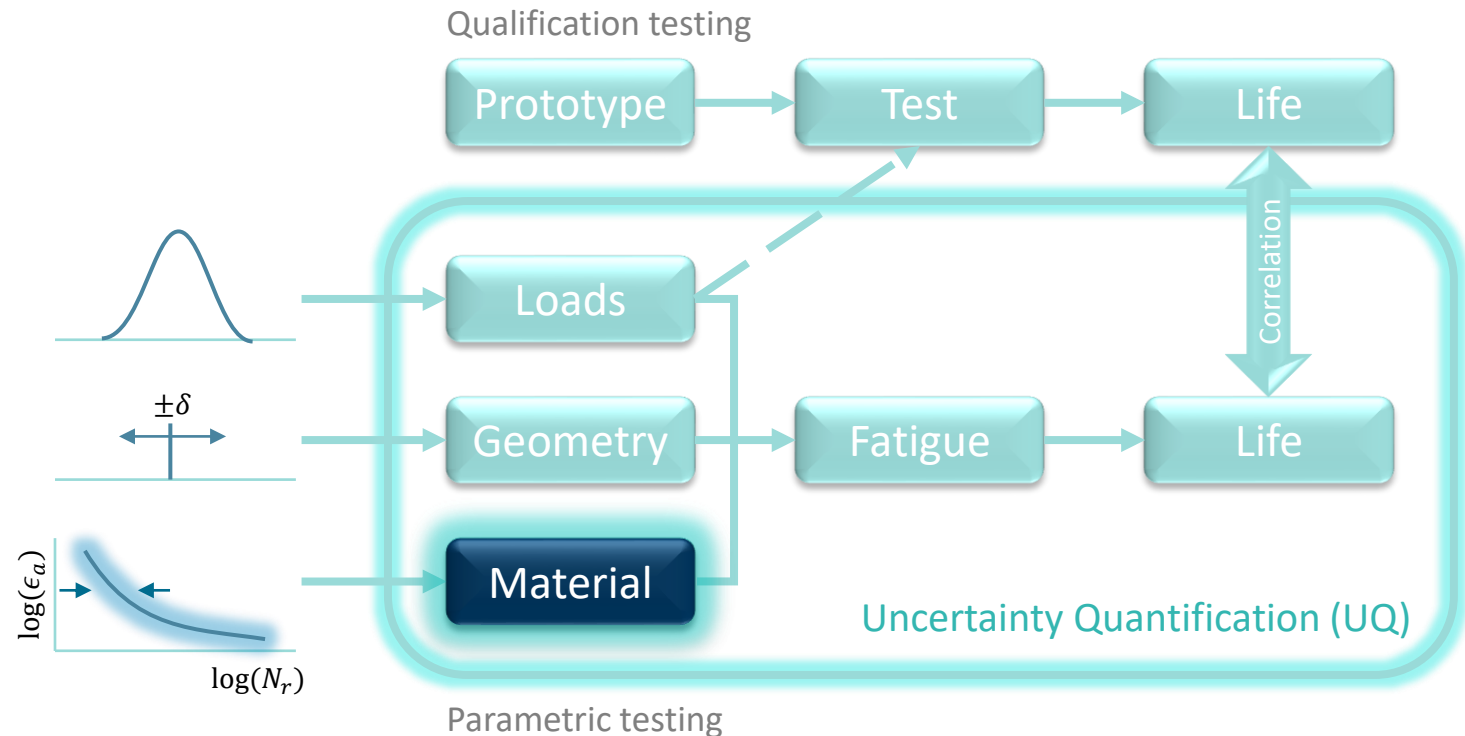


Challenge:

- How to validate requirement is met considering high degree of uncertainty implicit in fatigue ageing?

Solution:

- We must simulate the uncertainty too.



Agenda

1. **Strength** – Monotonic tensile parameters

- Strength & ductility
- Non-linear Isotropic hardening
- Uncertainty

2. **Ageing** – Cyclic & fatigue parameters

- Strain-controlled testing
- Non-linear Kinematic hardening
- Strain-Life (EN) fatigue
- Uncertainty

3. Today's challenges

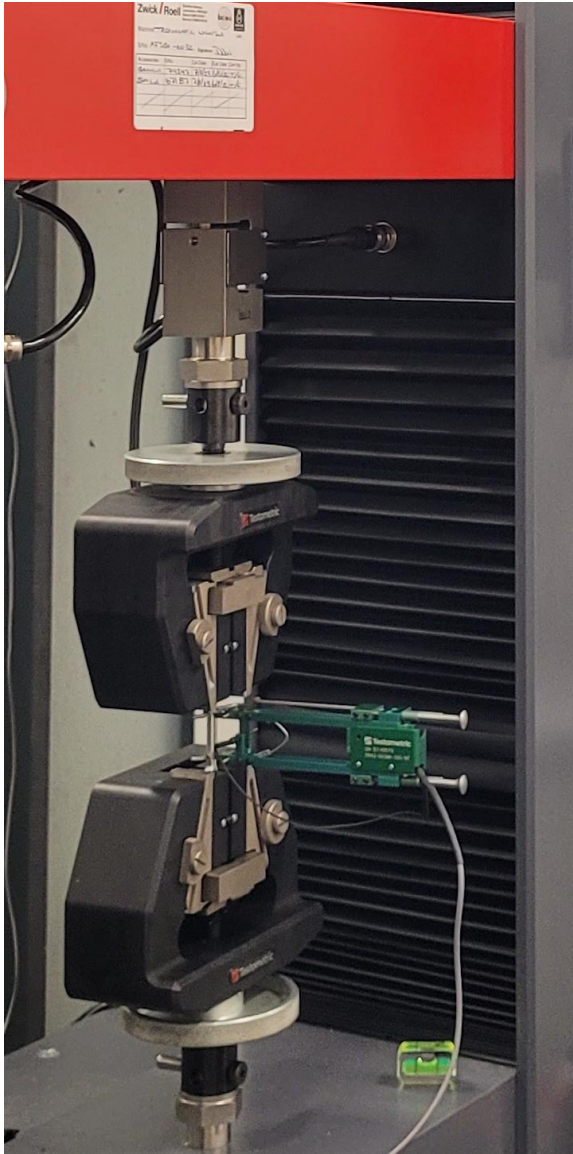
New IMechE student members

*Alex – Graduate Degree-level apprentice.
Just starting his MSc in Materials Science.*



Rebecca – new Degree-level apprentice.

Monotonic Tensile Parameters



Job reference: C12345

Test date: 04/10/2022

Material: Sample Material

Engineer: AP

Condition: As Received

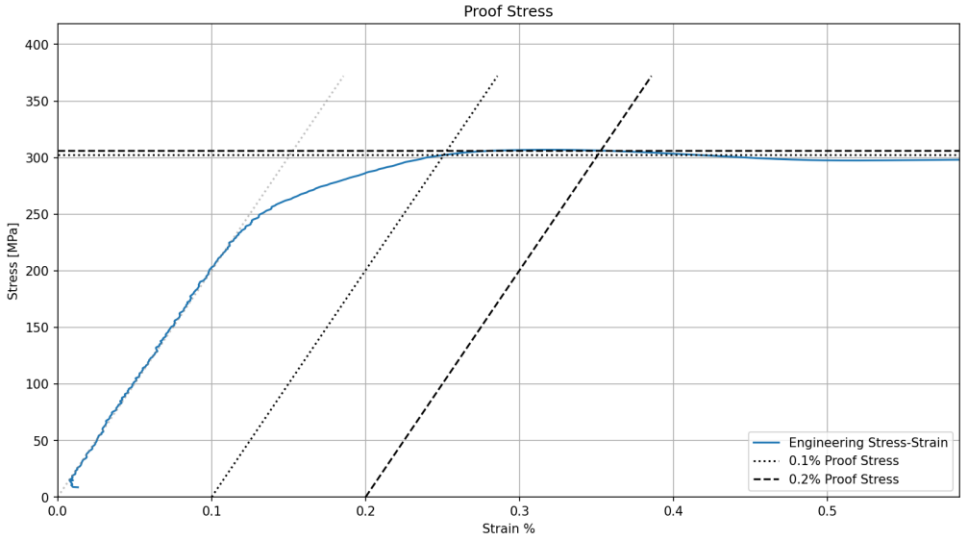
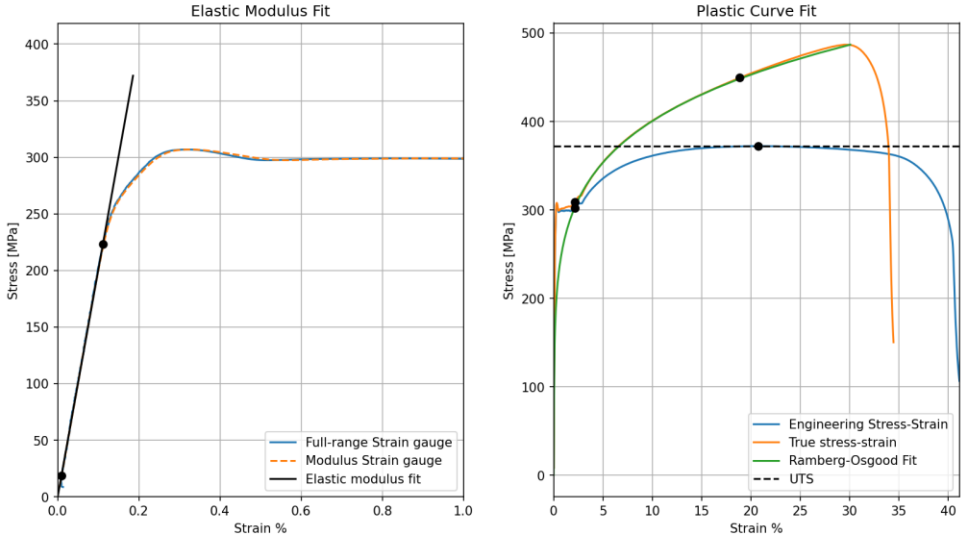
Test number: T02

Software version: 3.0.0

Customer: HBK AMCT

Test rig / temp: Kestrel (W+B) at Room Temperature

Elastic strain rate: 0.00274 % strain/s



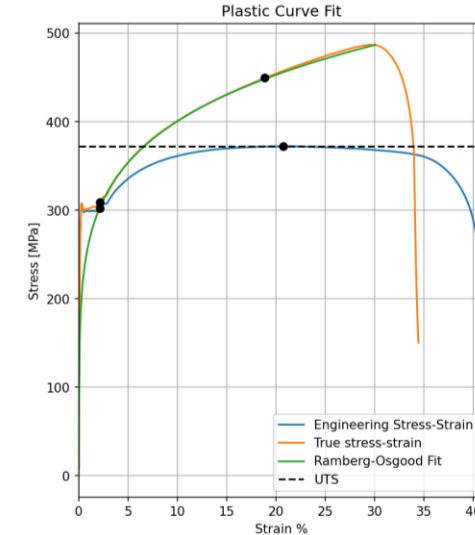
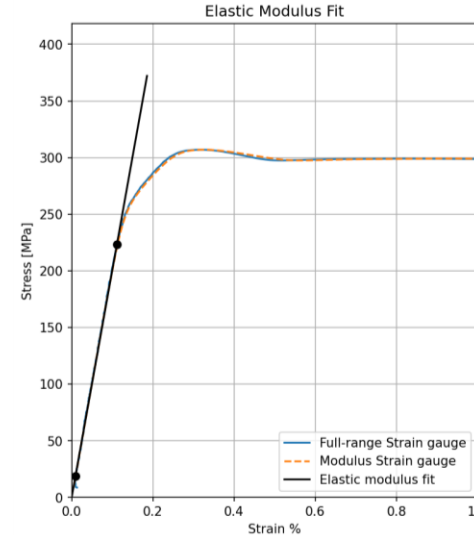
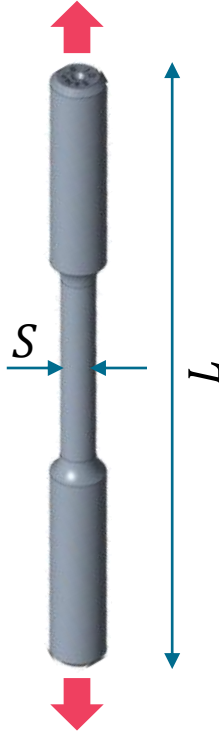
E	E_c	σ_u	$\sigma_{0.1\%}$	$\sigma_{0.2\%}$	K	n	$A\%$	$Z\%$
200,459.3 MPa	197,962.6 MPa	372.1 MPa	302.2 MPa	306.1 MPa	601.8 MPa	0.1754	41.43 %	80.75 %

Monotonic Tensile Parameters – Strength & Ductility

Strength:

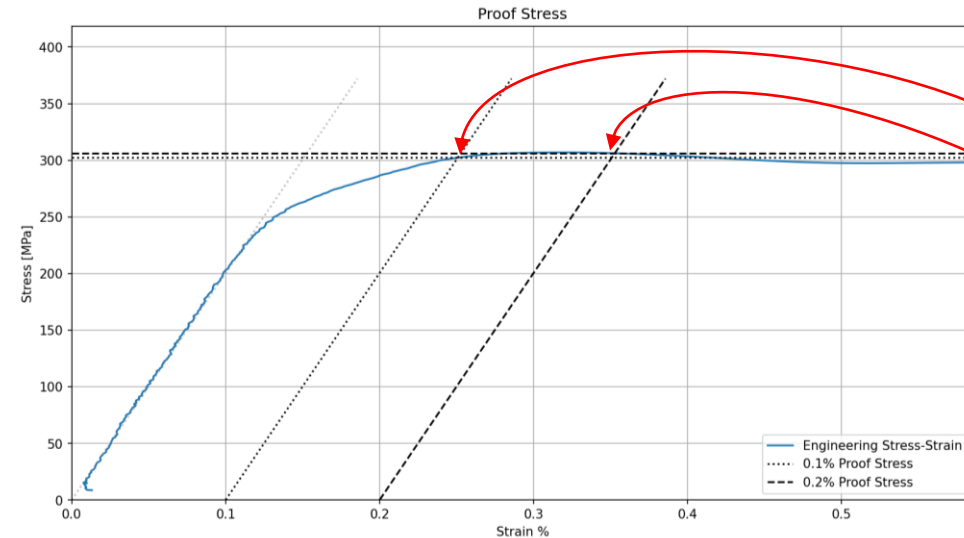
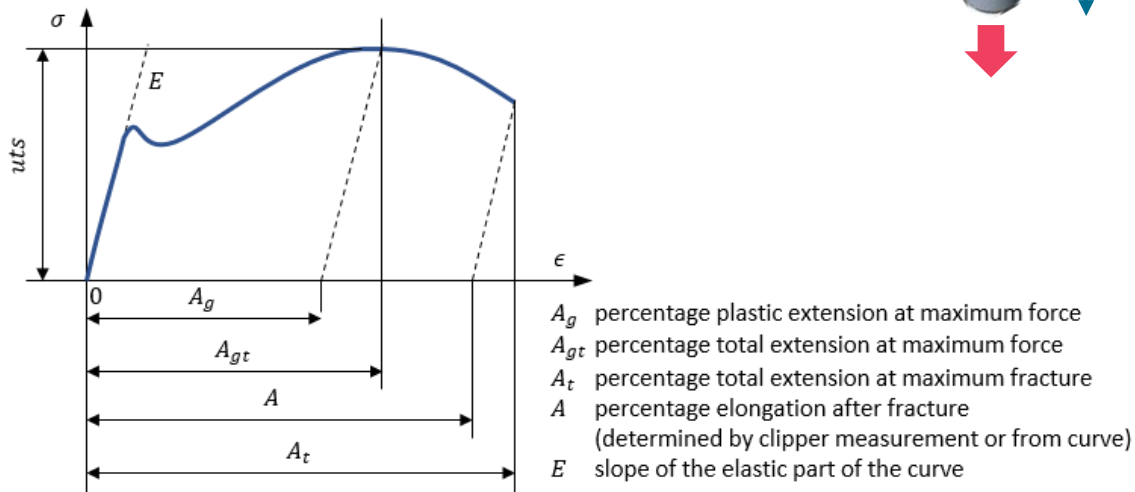
Ductility – from specimen dimensions:

- $A = \frac{L_u - L_o}{L_o}$: elongation %
 L_o : original gauge length
 L_u : final gauge length after fracture
- $Z = \frac{S_o - S_u}{S_o}$: reduction in area %
 S_o : original cross-sectional area
 S_u : final cross-sectional area after fracture



σ_u : Ultimate Tensile Strength

Ductility – from measurement curves:



Proof strength:

- $\sigma_{0.1\%}$
- $\sigma_{0.2\%}$

Monotonic Tensile Parameters – Isotropic Hardening

Ramberg-Osgood (RO) isotropic hardening model:

$$\epsilon_T = \underbrace{\frac{\sigma_T}{E}}_{\text{Elastic}} + \underbrace{\left(\frac{\sigma_T}{K}\right)^{\frac{1}{n}}}_{\text{Plastic}}$$

Elastic:

- E obtained through linear regression of σ and ϵ
- $E_c = \frac{\Delta\sigma}{\Delta\epsilon}$: chord modulus between 2 points

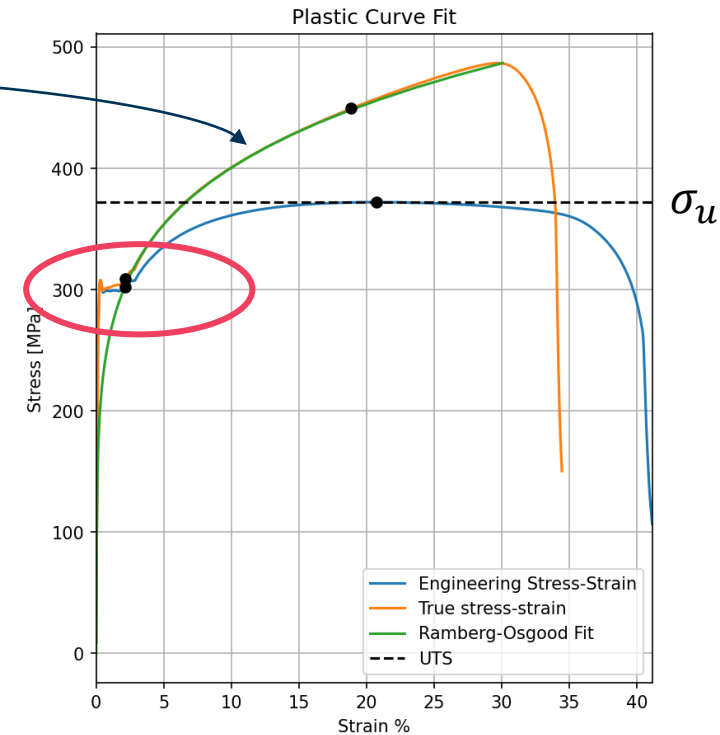
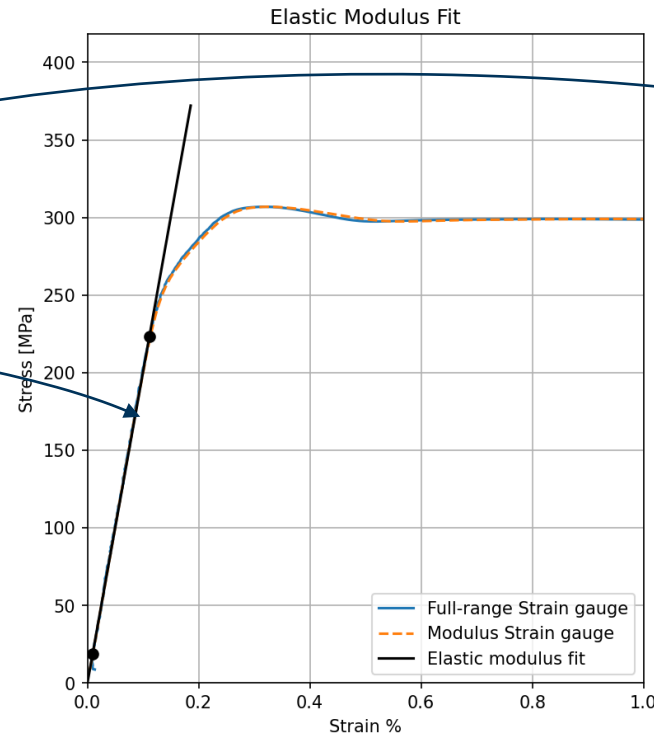
Plastic:

- RO curve based on log regression through *True Stress* σ_T vs. plastic *True Strain* ϵ_T

$$\sigma_T = \sigma(1 + \epsilon)$$

$$\epsilon_T = \log(1 + \epsilon)$$

$$\epsilon_{T_{plastic}} = \log\left((1 + \epsilon) - \frac{\sigma}{E}\right)$$



Other Isotropic hardening models:

- Chaboche
- Piecewise linear interpolation

Monotonic Tensile Parameters – *Uncertainty*

Min 3x as received:

- Used for quality control of supplied material before fatigue testing.

Min 3x cyclically stabilised:

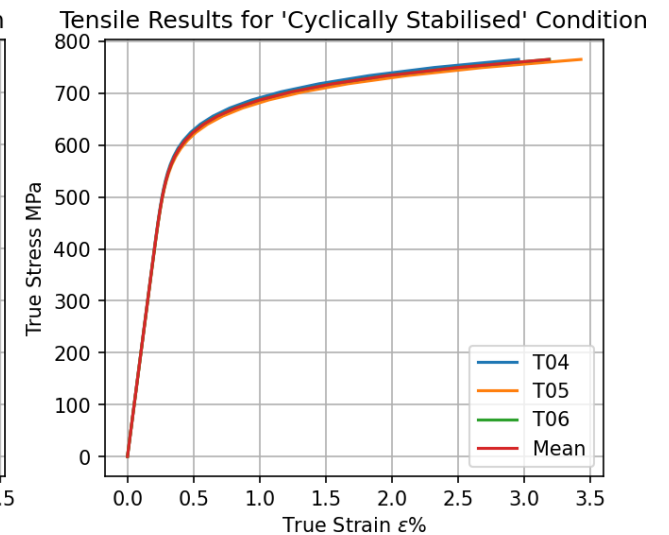
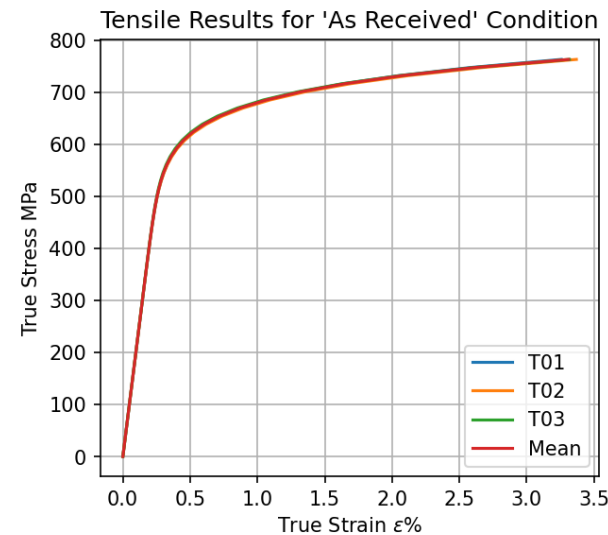
- Cycled for ~100 – 2000 cycles in strain control close to plastic limit prior to tensile testing.
- Used for cyclically stabilised static parameters in fatigue analysis.

Uncertainty on fatigue life relatively insensitive to uncertainty on tensile parameter, $\therefore$ 3 tests usually satisfactory for fatigue analysis.

But, if ultimate strength is a design-limiting factor, then suggest full statistical analysis based on ≥ 15 specimens.

	T04	T05	T06	Mean
Strength	UTS	770.0	761.0	764.7
	Elastic modulus E	197,635	196,240	196,030
	0.1% proof stress	679.3	664.1	674.4
	0.2% proof stress	688.1	675.2	682.4
Plasticity	K	1,001.3	993.2	995.0
	n	0.0734	0.0746	0.0738
Ductility	Elongation A%	24.4%	25.6%	24.9%
	Reduction of area Z%	72.1%	74.1%	72.2%

Parameters for elastic and elastic-plastic analysis with isotropic hardening



Agenda

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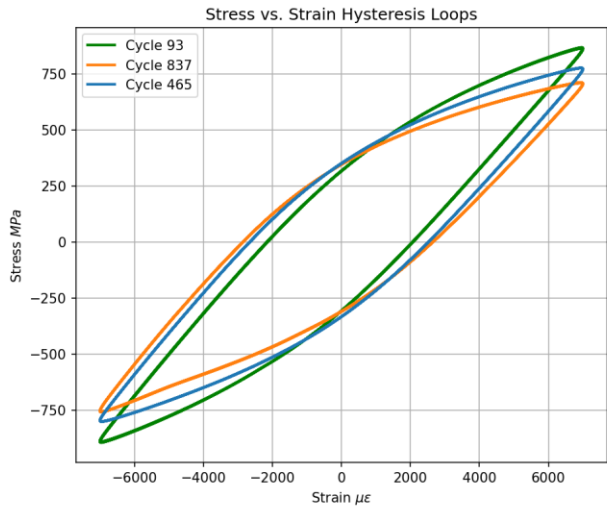
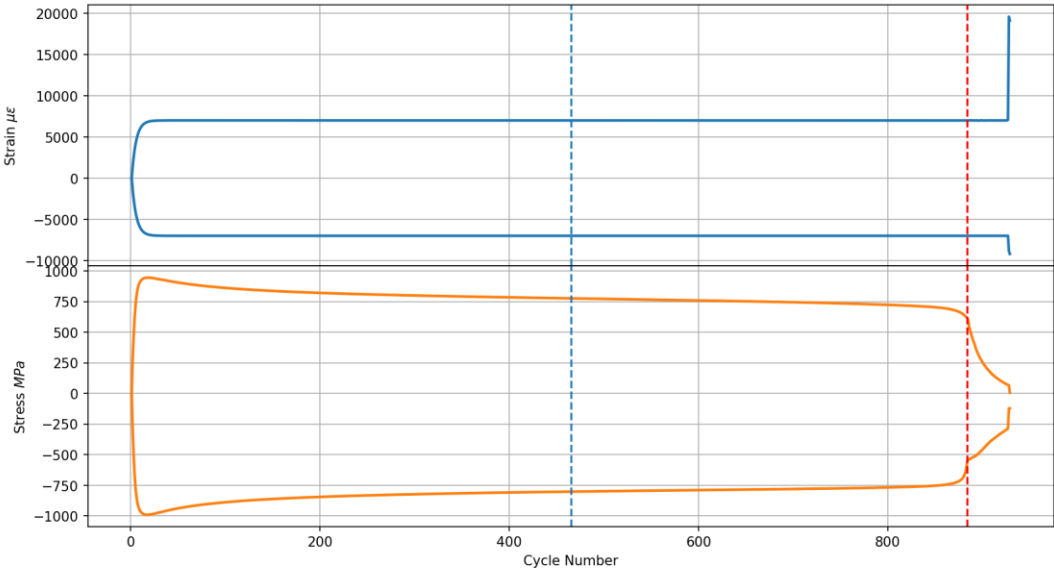
Cyclic & Fatigue Parameters



Strain Controlled Fatigue Test Report



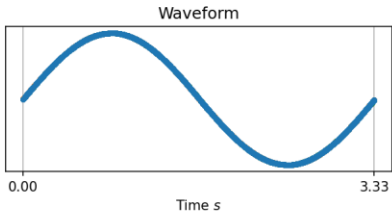
Job reference:	C12345	Material:	Sample Material
Test number:	11	Customer:	HBK AMCT
Test date:	21/10/2020	Engineer:	MC
Software version:	3.0.1	Test rig:	Emily (Tiab)



Cross sectional area: 16.100 mm²

Test temperature: Ambient
Test frequency: 0.3 Hz
Stable loop: 465

Cycles to separation: 929
5% load drop: 740
10% load drop: 858
20% load drop: 884
50% load drop: 893



Cycle No.	Total Stress	Total Strain	Elastic Strain	Plastic Strain
93	1758.5 MPa	14000	9780	4220
465	1577.1 MPa	13999	8817	5182
837	1468.4 MPa	14000	8496	5504

Cyclic & Fatigue Parameters

Job reference: C12345

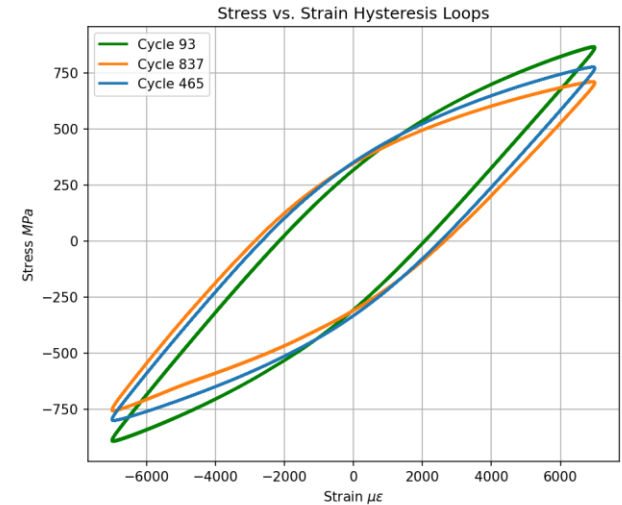
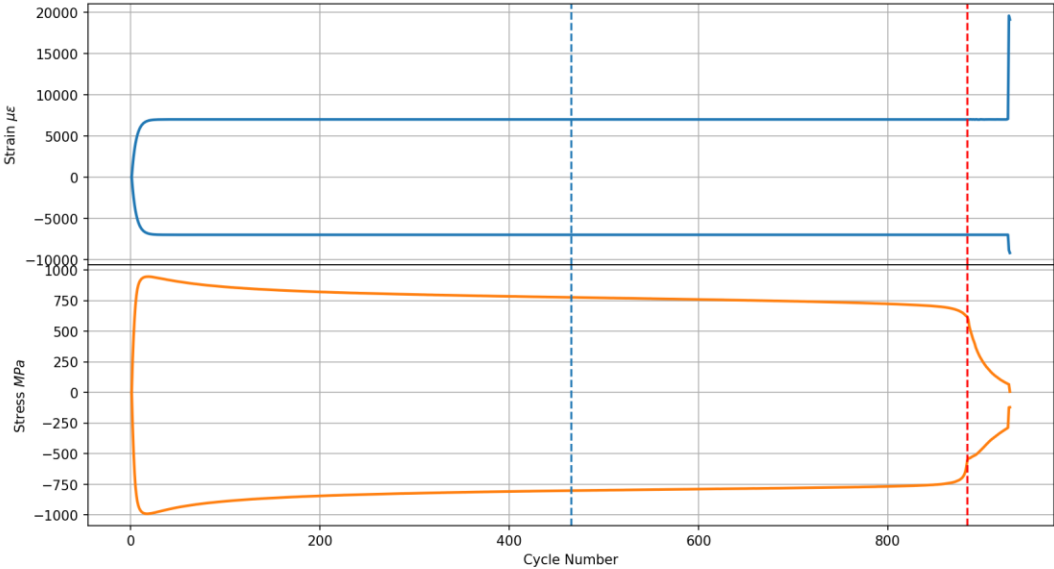
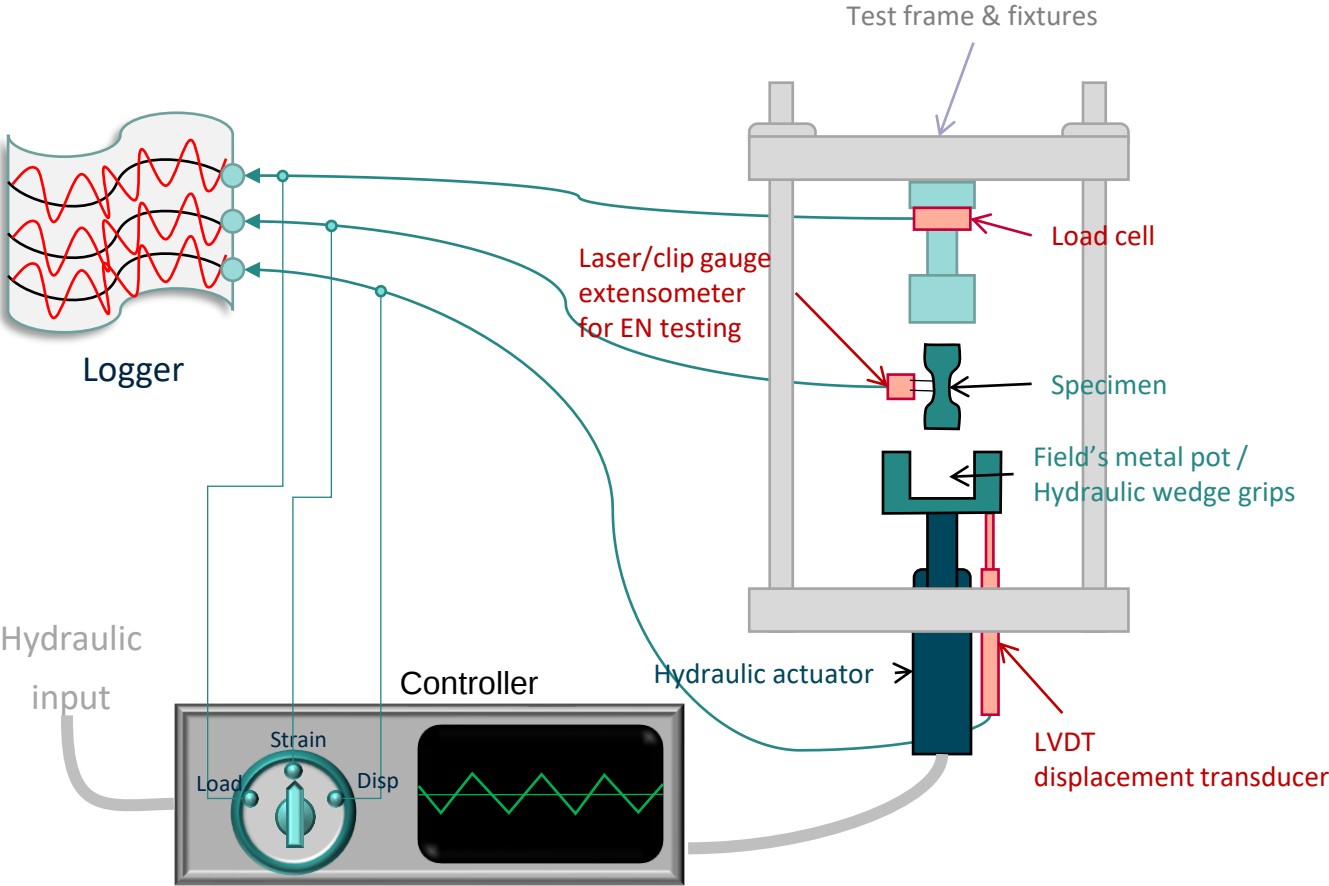
Test number: 11

Test date: 21/10/2020

Software version: 3.0.1

Material: Customer: Engineer: Test rig:

Sample Material: HBK AMCT MC Emily (Tiab)



Cross sectional area: 16.100 mm²

Test temperature: Ambient

Test frequency: 0.3 Hz

Stable loop: 465

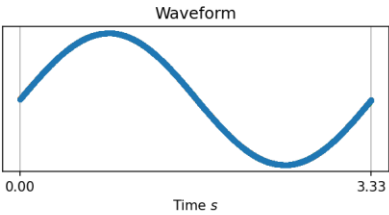
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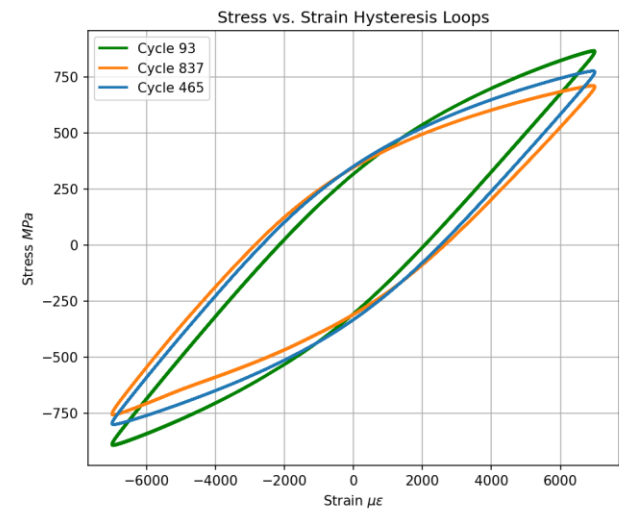
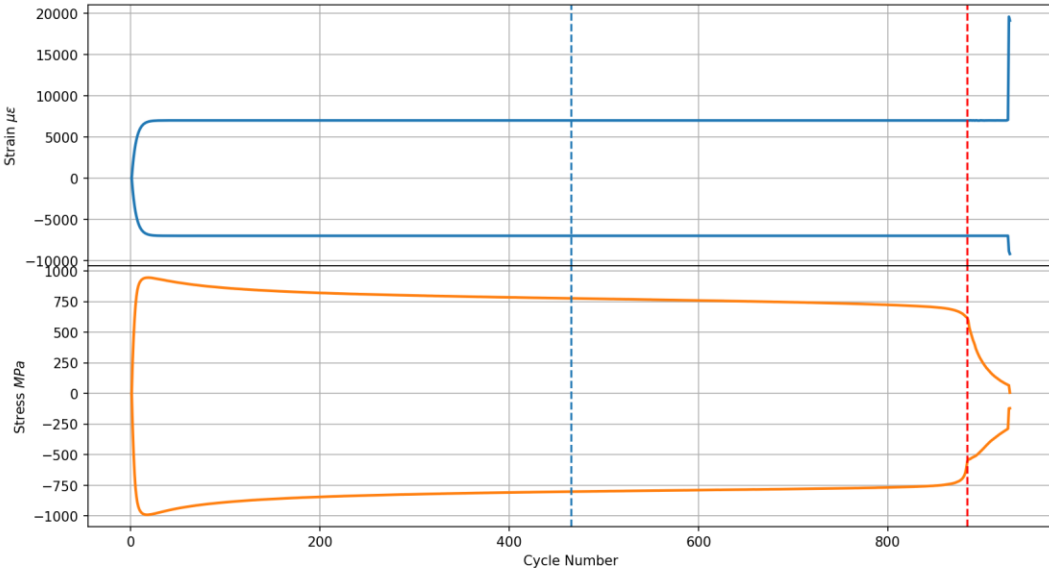
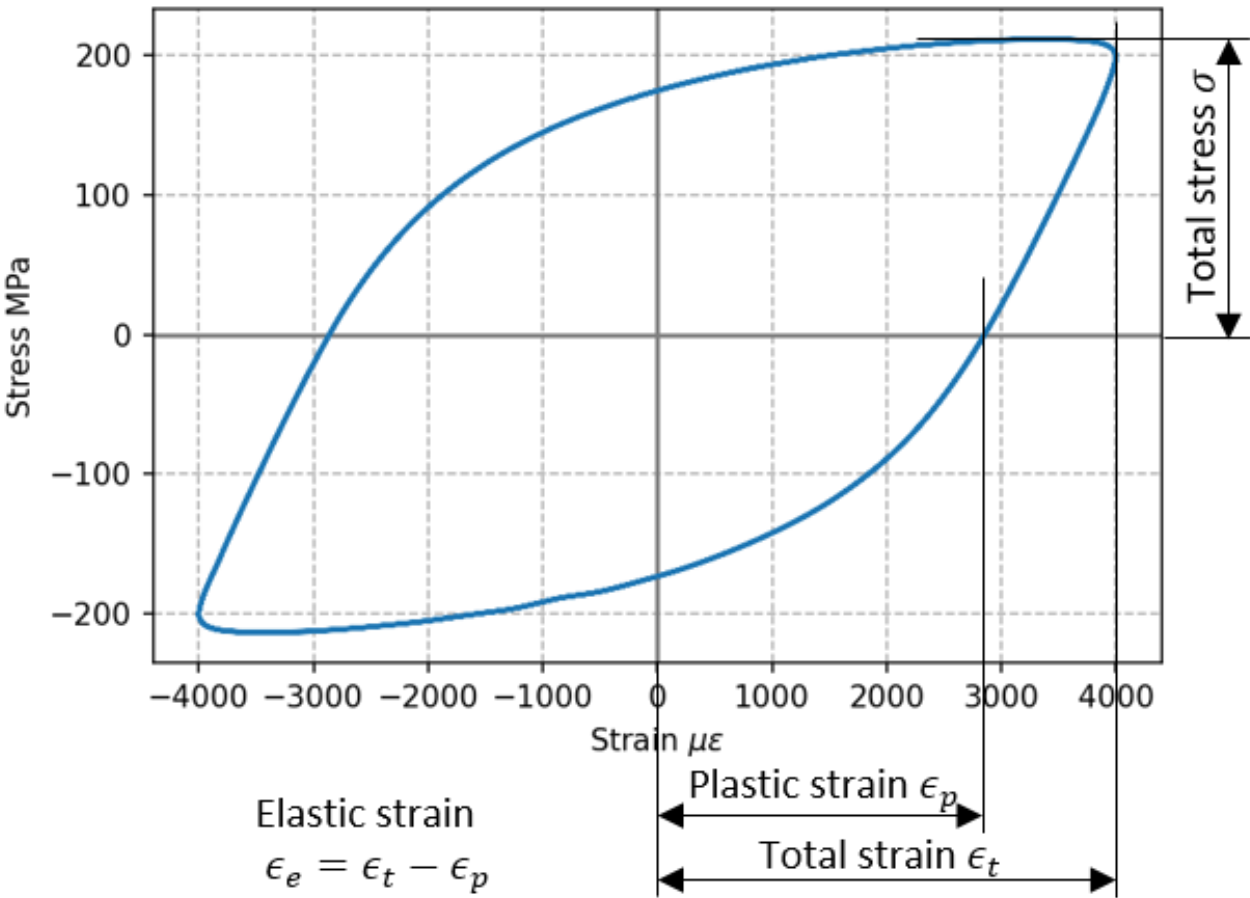
50% load drop: 893



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Cyclic & Fatigue Parameters

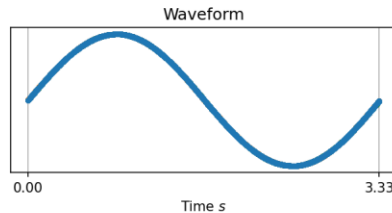
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Test temperature: Ambient
Test frequency: 0.3 Hz
Stable loop: 465

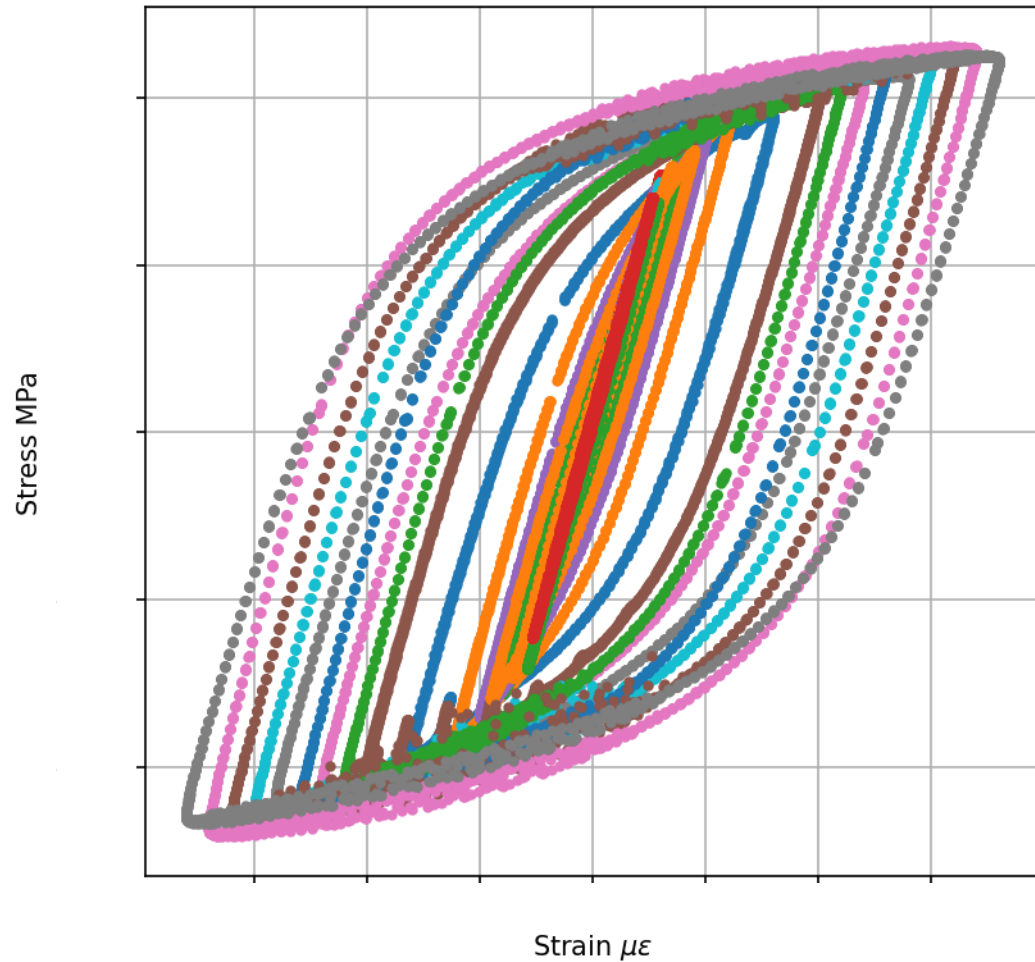
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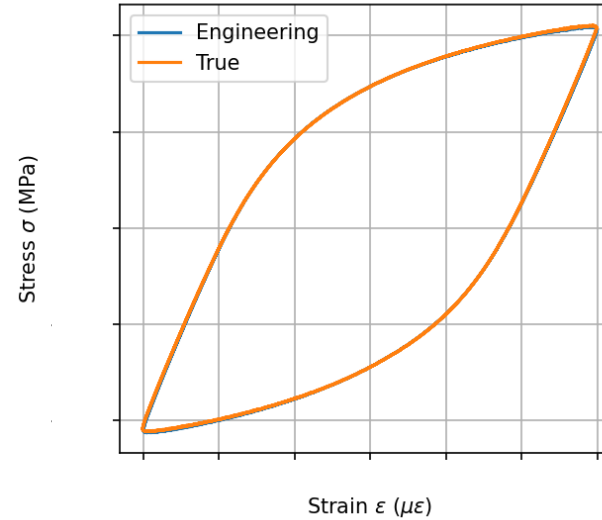
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Cyclic & Fatigue Parameters – Kinematic hardening

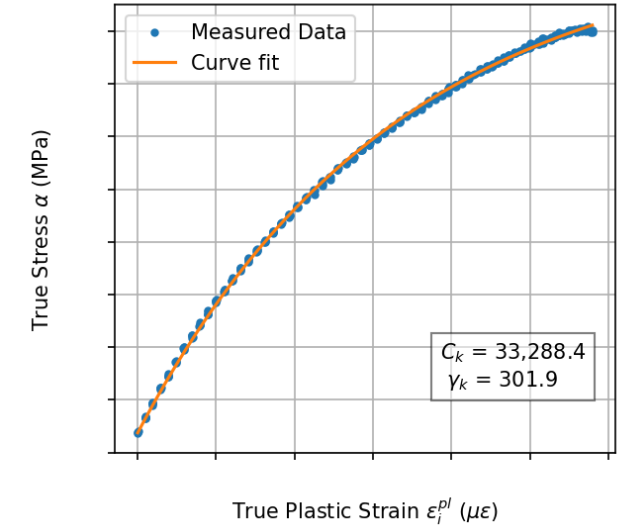
Raw Stable Loops



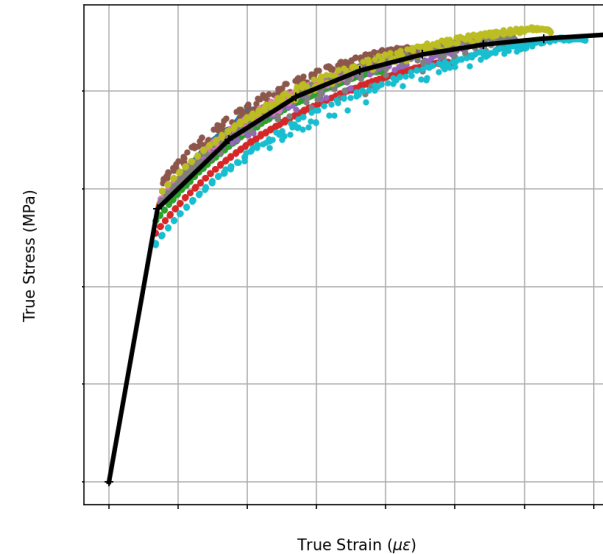
a) Stabilised Loop, Test F07, Cycle 1405



b) Chaboche Curve Fitting



b) Chaboche Overall Curve Fit



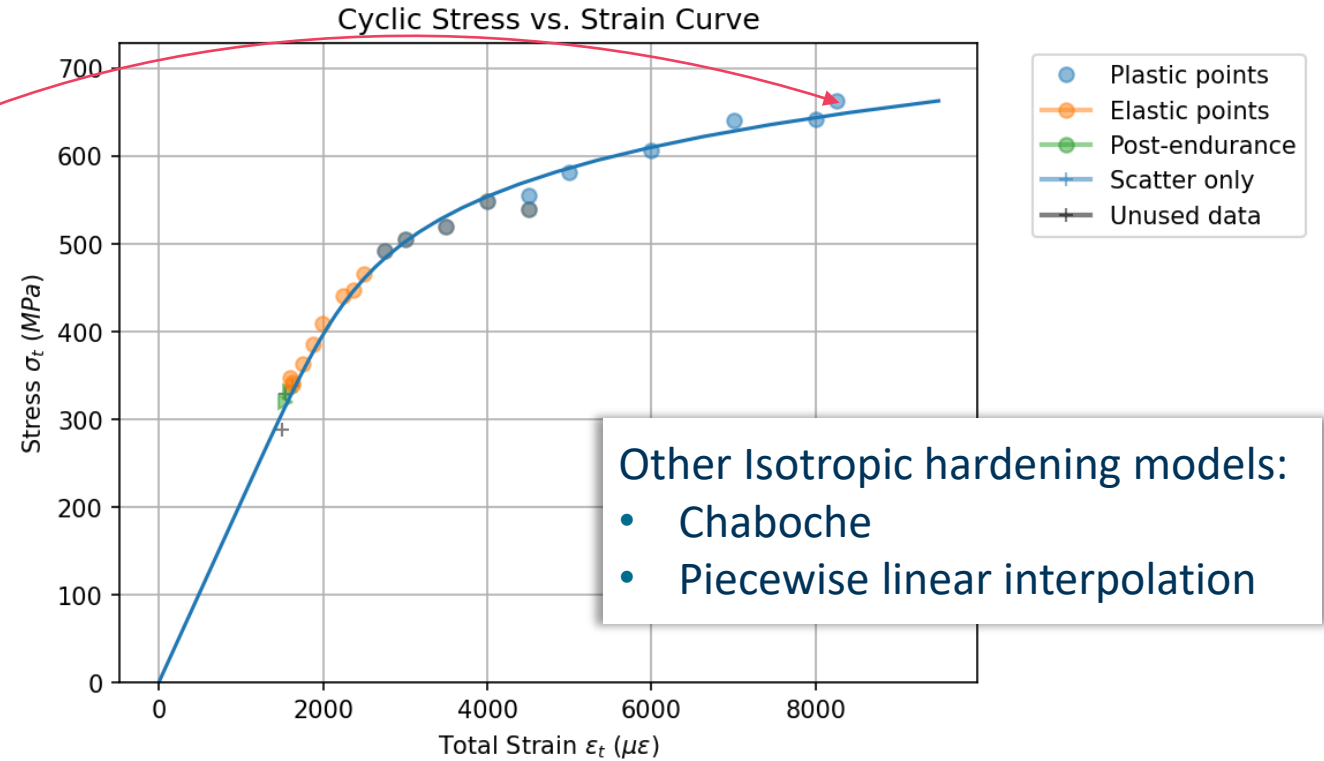
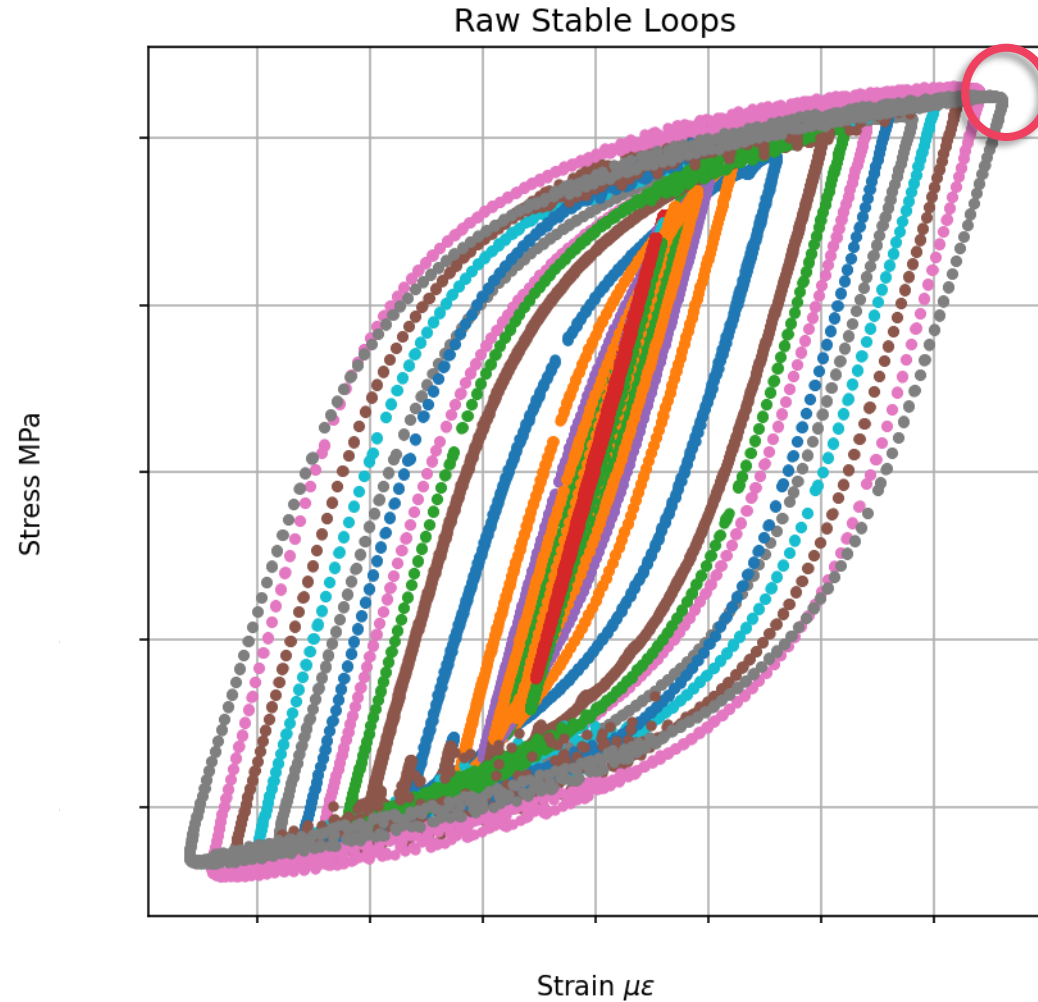
Kinematic hardening models:

- Ramberg-Osgood-Neuber
- Chaboche

Challenge:

- How to determine an average curve

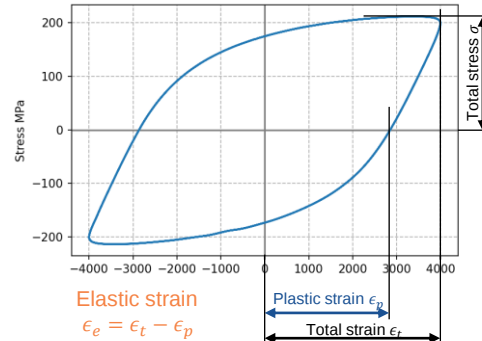
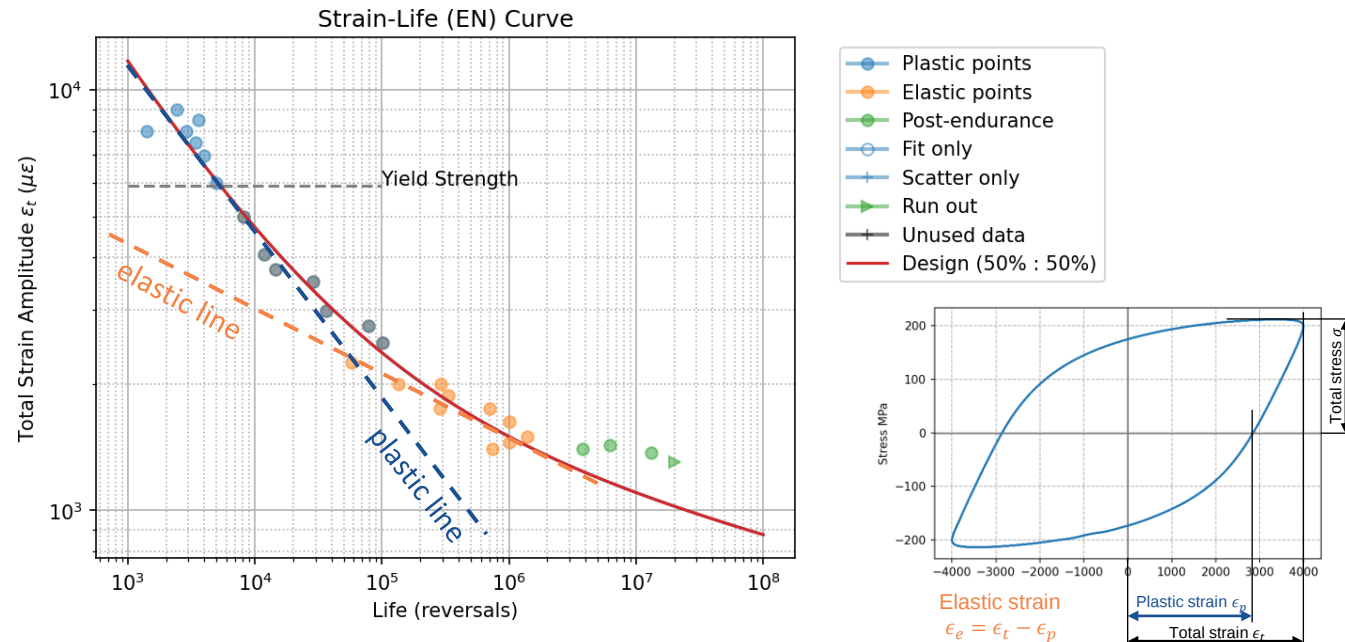
Cyclic & Fatigue Parameters – Kinematic hardening



Plot peak Total stress σ vs. peak Total strain ϵ_t for all tests, and fit Ramberg-Osgood equation

$$\epsilon = \frac{\sigma}{E} + \left(\frac{\sigma}{K'} \right)^{\frac{1}{n'}}$$

Cyclic & Fatigue Parameters – EN characterisation analysis



1. Plot Total strain amplitude ϵ_t vs. reversals to failure N_r for the specified load-drop.

2. Fit **plastic line** using plastic strain ϵ_p

$$\epsilon_p = \epsilon_f' N_r^c$$

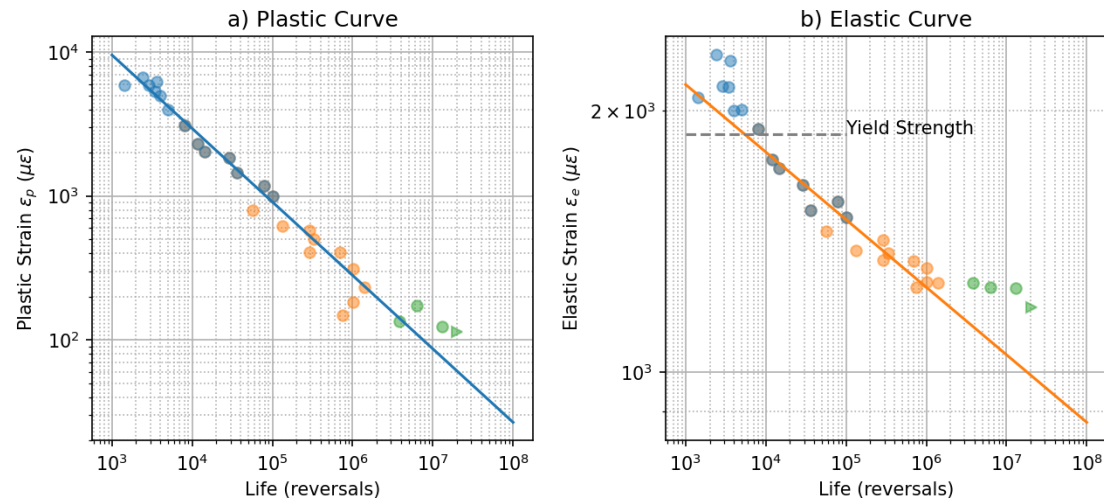
3. Fit **elastic line** using elastic strain

$$\epsilon_e = \frac{\sigma_f'}{E} N_r^b$$

4. Parameter re-optimisation

$$\epsilon_t = \frac{\sigma_f'}{E} N_r^b + \epsilon_f' N_r^c$$

5. Outlier analysis

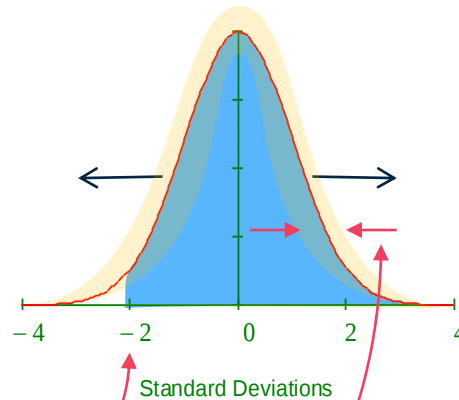


Cyclic & Fatigue Parameters – Uncertainty Quantification

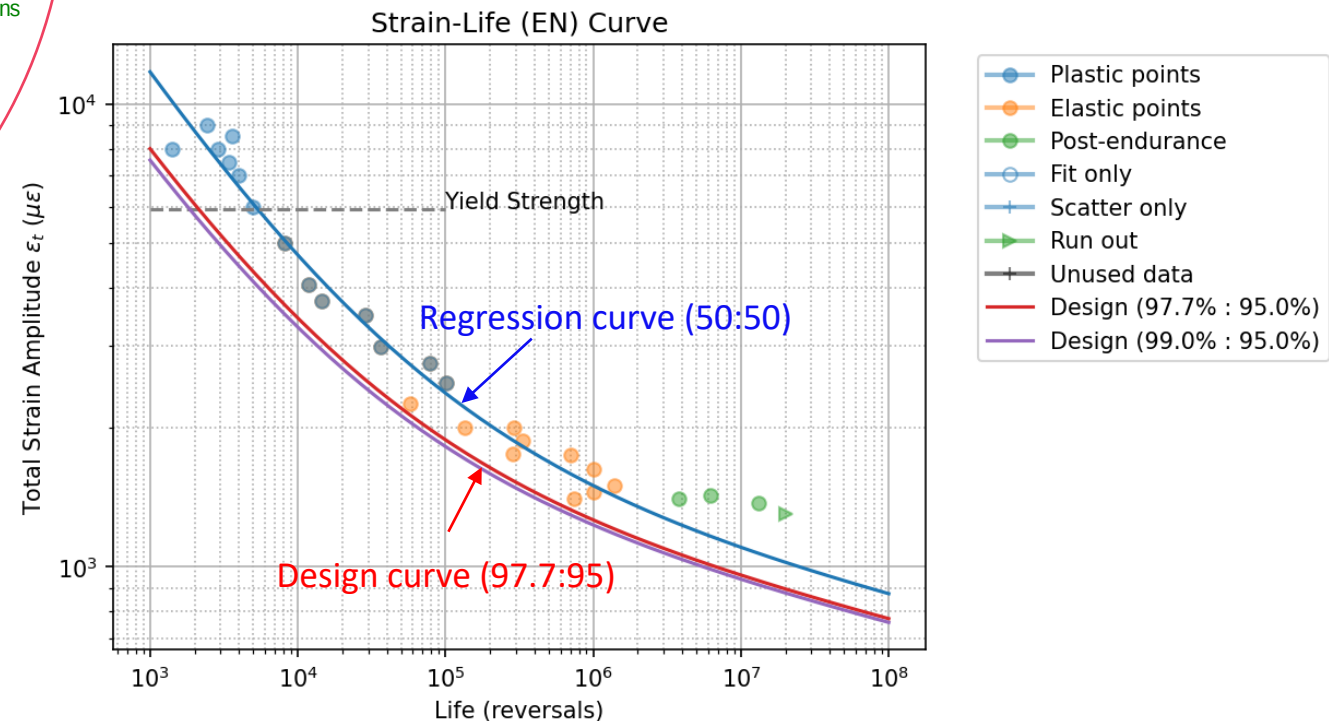
Deterministic design approach:

Design curve(reliability % : confidence %)

- Reliability statistic:
(acceptable number of failures)
 - 50% (mean expected life)
 - 95% (5 failures in 100)
 - 97.7% (2 sigma)
 - 99% (1 failure in 100)
- Confidence statistic:
(sensitivity to sample size)
 - 50% (mean expected life)
 - 90 – 95% (typical design curves)



For a **duty cycle** of 10 years / 150,000 miles,
I require **reliability** exceeding 95%
(i.e. fewer than 5% failures),
with a **confidence level** of at least 90%

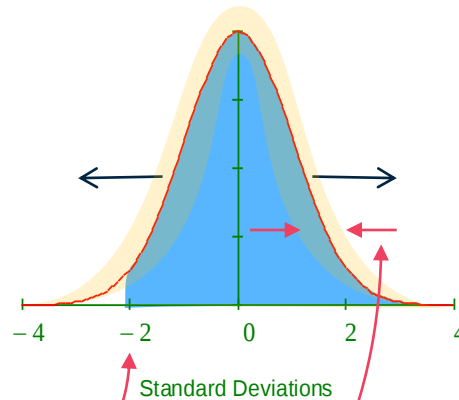


Cyclic & Fatigue Parameters – Uncertainty Quantification

Deterministic design approach:

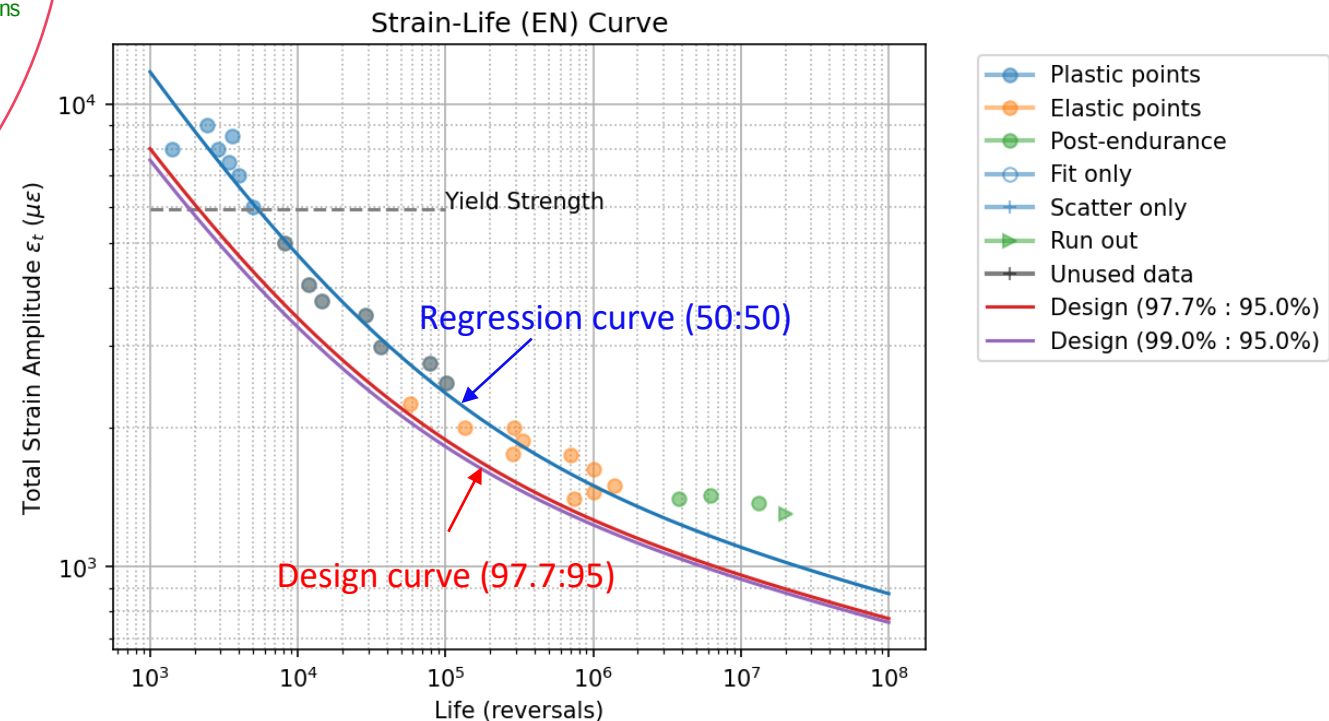
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$$\epsilon_t = \frac{\sigma'_f}{E} 10^{k_{Ue} \cdot SE_e N^b} + \epsilon'_f 10^{k_{Up} \cdot SE_p N^c}$$

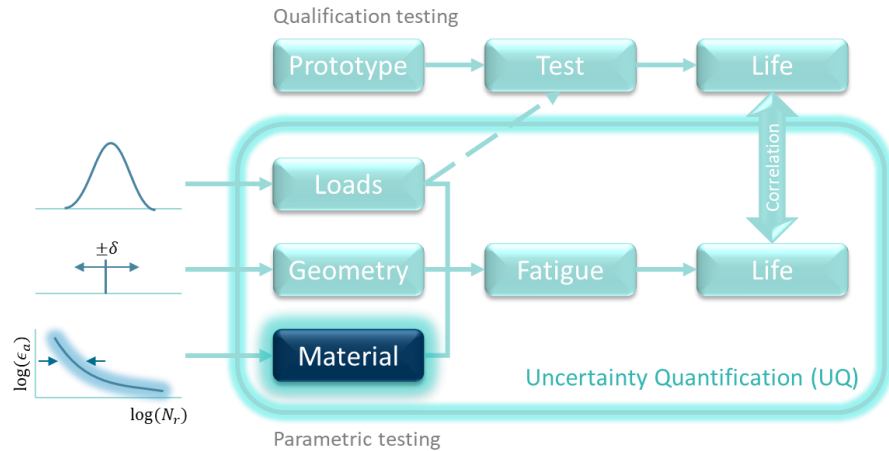
$$k_U = \frac{z_\rho + \frac{z_\gamma}{\sqrt{n}} \sqrt{1 - \frac{z_\gamma^2}{2(n-1)} + \frac{n z_\rho^2}{2(n-1)}}}{1 - \frac{z_\gamma^2}{2(n-1)}}$$



Cyclic & Fatigue Parameters – *Uncertainty Quantification*

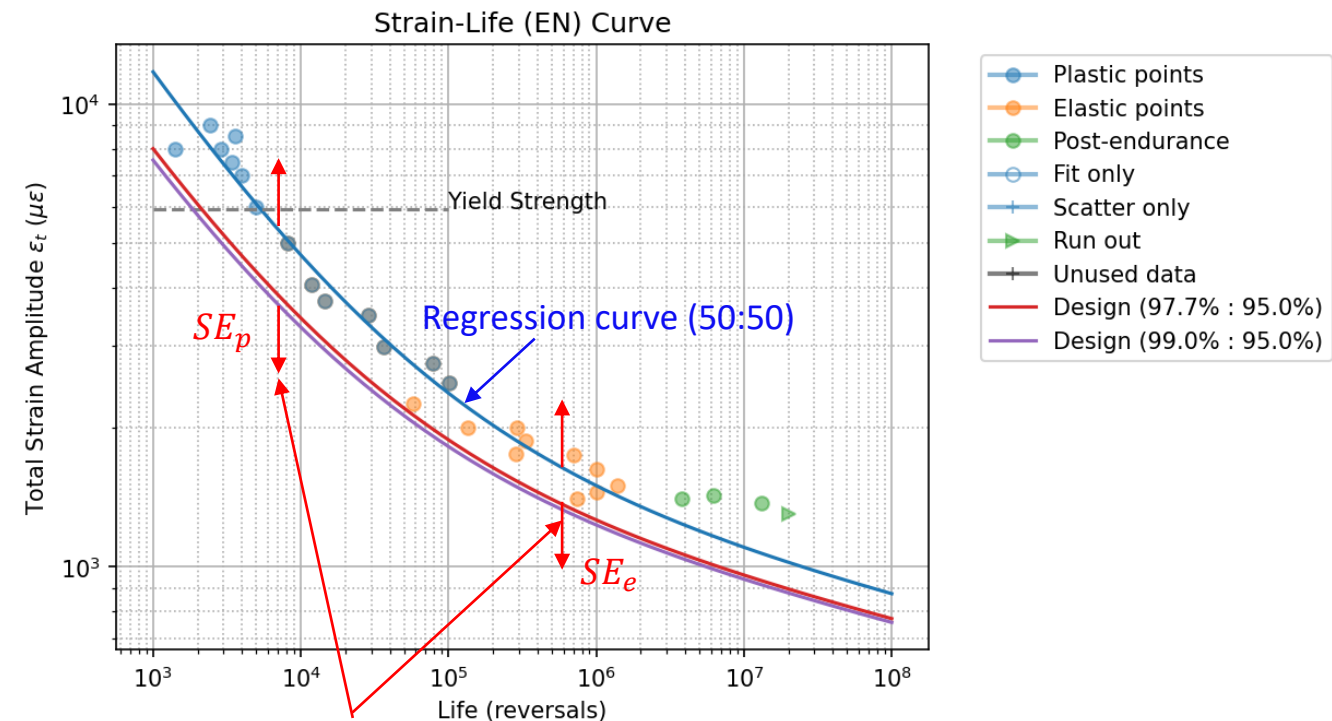
Stochastic design approach:

- Montecarlo uncertainty simulation using standard error values.



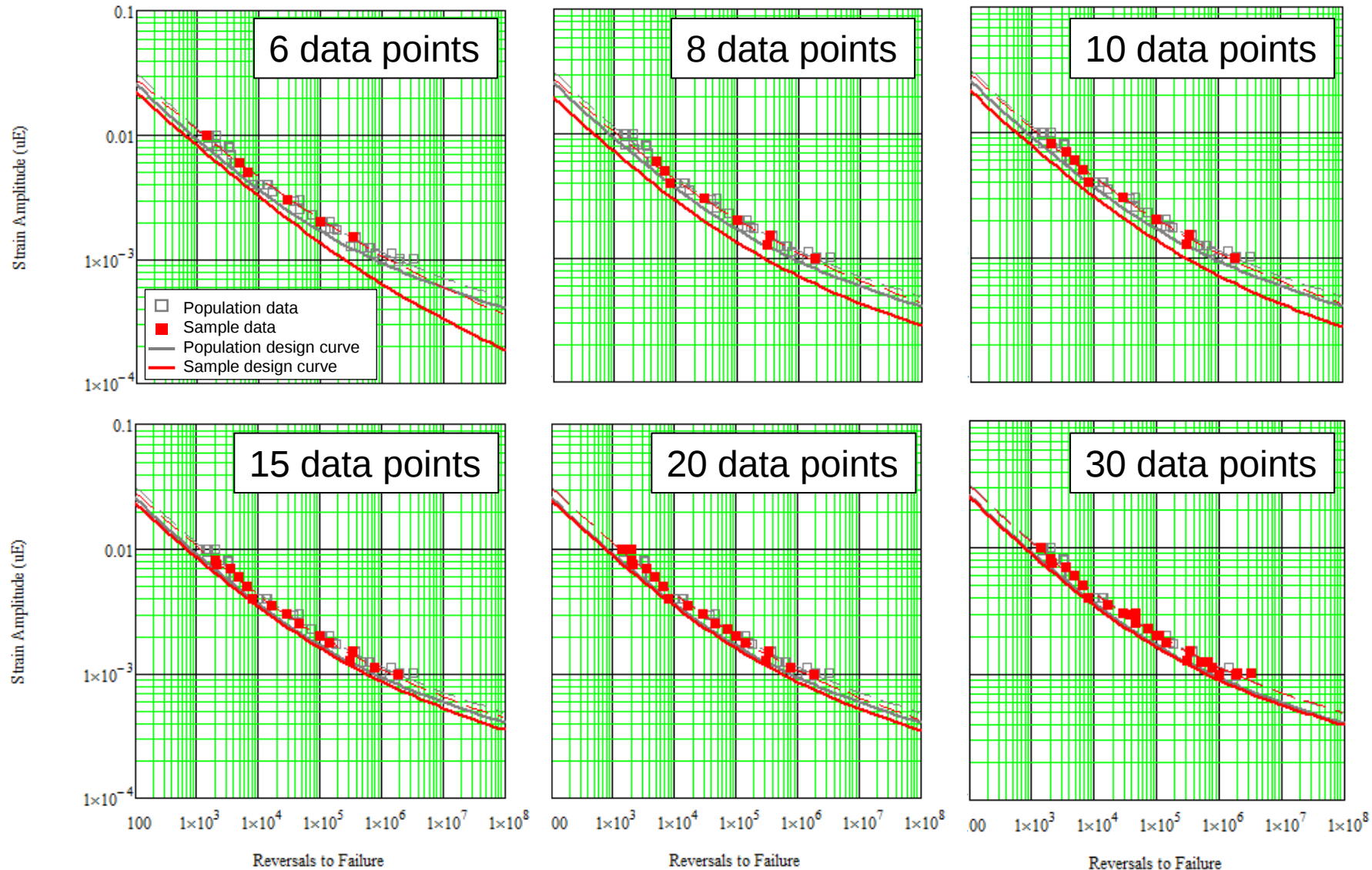
$$\epsilon_t = \frac{\sigma'_f}{E} 10^{u \cdot SE_e} N^b + \epsilon'_f 10^{u \cdot SE_p} N^c$$

u is the random number of standard deviations
(DesignLife %age Certainty of Survival)



Plastic & elastic lines have different SE values (SE_p and SE_e)

Cyclic & Fatigue Parameters – *Uncertainty Quantification*



6 – 10 samples

- Inadequate for EN
- Absolute minimum for SN

15 - 20 samples

- Absolute minimum for EN
- Usually adequate for SN

20 - 30 samples

- Good quality SN & EN design curve (97.5 : 95)

Agenda

1. Strength – Monotonic tensile parameters

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3. Today's challenges

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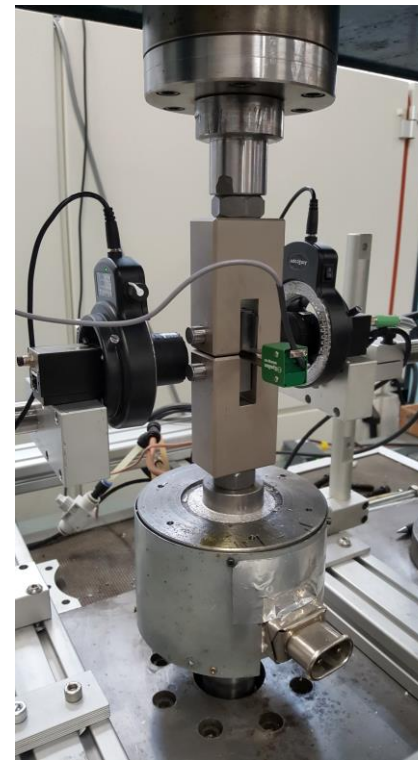
Challenges – *Driven by Environmental Sustainment*

▲ New high-performance, light-weight materials:

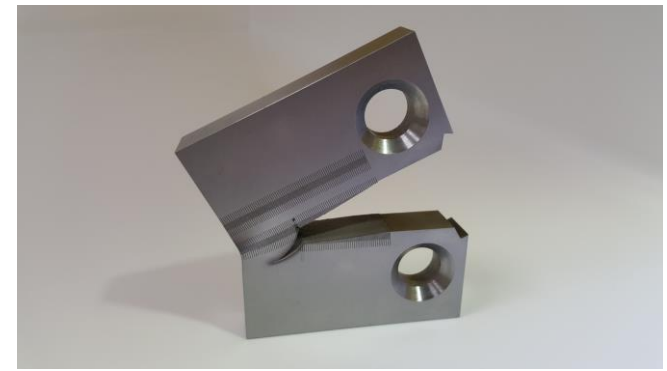
- Wider range of materials:
'exotic' metals, polymers, composites, laminates...
- Jointing of dissimilar materials
 - Continuous joints: *welds, adhesives...*
 - Discrete fasteners: *spot welds, rivets, SPR, adhesive hybrids...*
- New production techniques:
Additive Manufacturing, sintering, casting...

▲ Improved materials models to reduce epistemic uncertainties:

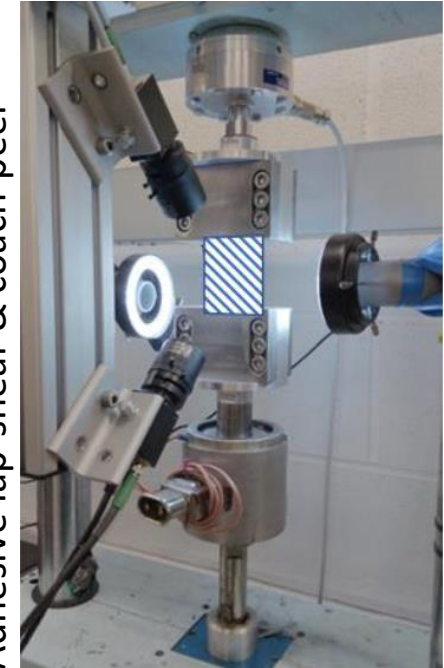
- R ratio (mean stress sensitivity) modelling
- Surface finish / treatment modelling
- High (& low) temperature, thermo-mechanical
- Reuse & life extension of ageing materials
- Load sequence effects modelling
- Crack growth and progressive failure modes



Crack growth & progressive failure modes



Thermo-mechanical fatigue

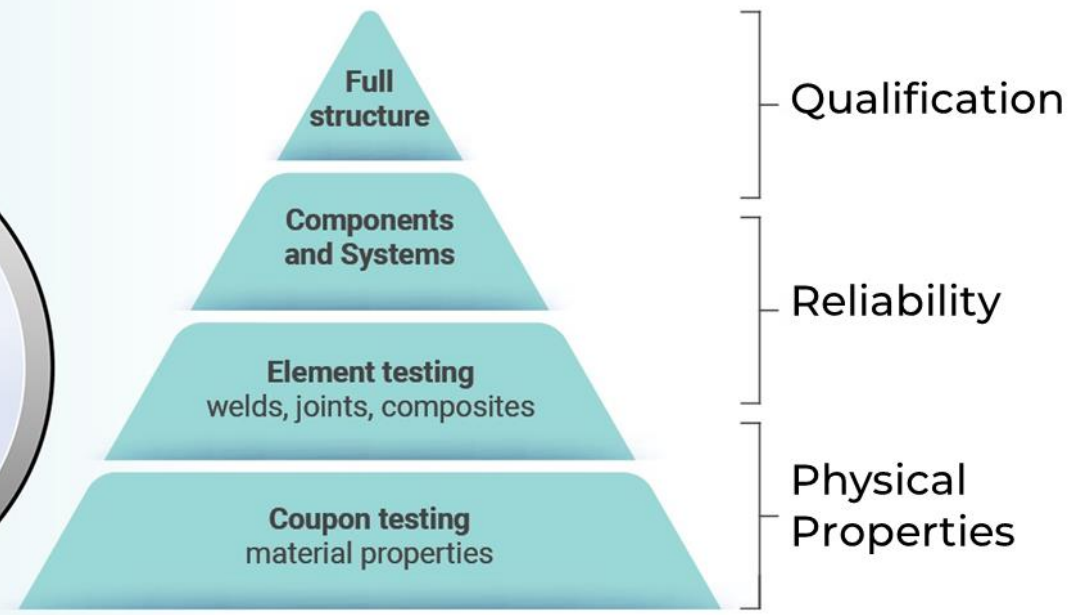
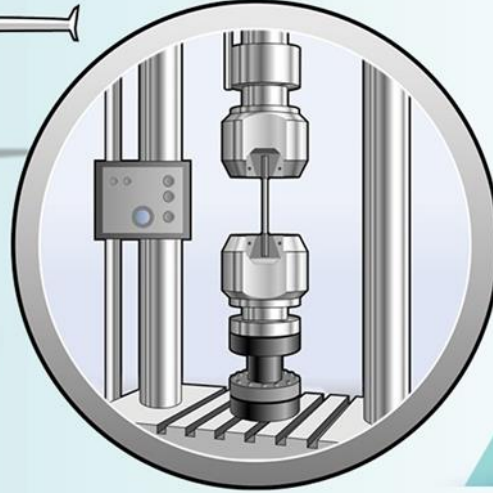
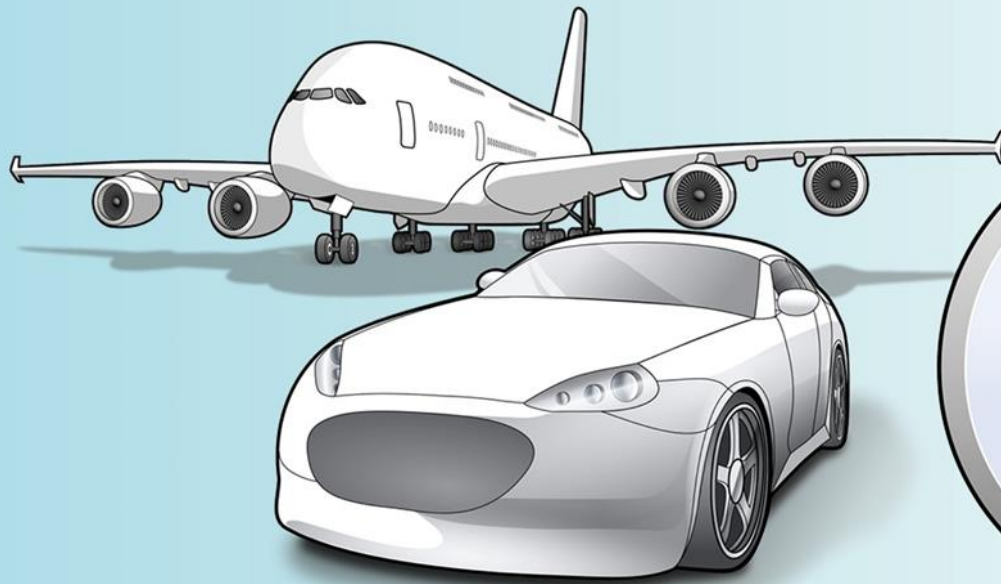


Adhesive lap-shear & coach-peel

Thank You

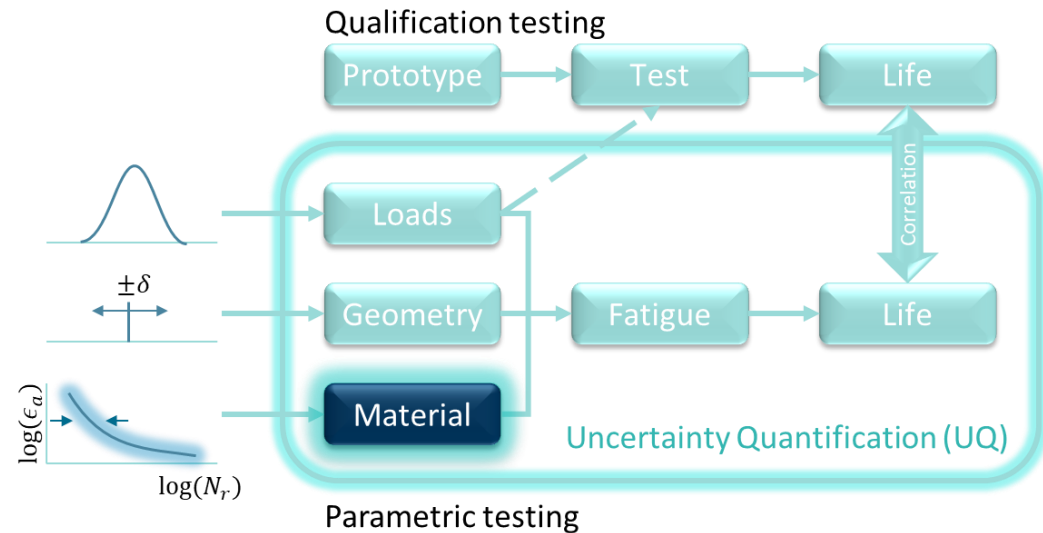
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2023 HBK Technology Days

14 November 2023



- Complementary material parametric testing and simulation for fatigue prediction

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